



Accelerated Insertion of Materials –

Managing Error and Uncertainty in Structures

**49th International Society for the
Advancement of Material and Process
Engineering Symposium and Exhibition**

17 May 2004

**S. Eric Cregger, Boeing
samuel.e.cregger@boeing.com**

**Anthony Caiazzo, Materials Sciences Corporation
Perry Pugliano and Raj Rajagopal, Boeing
Stan Uryasev, American Optimal Decisions**



Jointly accomplished by BOEING and the U.S. Government under
the guidance of NAVAIR.

Sponsored by DARPA/DSO and administered by NAVAIR through TIA N00421-01-3-0098



Approved for Public Release, Distribution Unlimited





Outline

- The Basic AIM-C Approach (as implemented)
 - Understand and Classify Potential Uncertainty Sources
 - Determine What's Important
 - Limit Uncertainty/Variation by Design and/or Process
 - Quantify Variation (Monte Carlo Simulation or Test)
- Data from Knowledge, Analysis, and Test
 - Previous Knowledge and Divergence Risk
 - Analysis
 - Test
 - Combined Data Approaches



Handling Uncertainty – The AIM-C Approach

- The First Step is Identifying and Understanding potential error sources
 - Maintains Visibility of potential errors
 - Forces step-by-step breakdown of the analysis/test process
 - Forces agreement on responses of interest
- Classifying them allows the team to determine appropriate strategies for addressing them.
- Types:
 - Aleatory Uncertainty (Variability, Stochastic Uncertainty)
 - Epistemic Uncertainty (Lack of Knowledge, e.g., unknown geometry)
 - Known Errors (e.g., mesh convergence, round-off error)
 - Unknown Errors (Mistakes, e.g. wrong material inputs used)



Identifying and Understanding Error Sources

| | Inherent variations associated with physical system or the environment (Aleatory uncertainty) Also known as variability, stochastic uncertainty E.G. manufacturing variations, loading | Uncertainty due to lack of knowledge (Epistemic uncertainty) inadequate physics models information from expert opinions. | Known Errors (acknowledged) e.g. round-off errors from machine arithmetic, mesh size errors, convergence errors, error propagation algorithm | Mistakes (unacknowledged errors) human errors e.g error in input/output, blunder in manufacturing |
|--|--|--|---|---|
| Lamina Stiffness/ Thermal Properties (CCA and/or Empirical) | Variation in all fiber and resin moduli, Poisson's ratio, and CTE properties. Test uncertainties such as specimen misalignment, load/displacement measurement | Unmeasurable Constituent Properties (transverse fiber modulus, etc.) Interphase effects | CCA: Use of model outside of bounds.(e.g., woven 3D preform) Empirical: Extrapolation beyond test data (fiber volumes, temperatures, etc.) | CCA: I/O errors, code bugs Empirical: Testing machine not calibrated. Poor specimen preparation; poor strain measurement techniques. |
| Laminate Stiffness Calculation (CLPT) | Variations in ply-thickness, ply angles, etc. | Assumes thin plate with no shear deformation, material or geometric nonlinearity, or significant transverse strains. | Use of model outside bounds for items listed under Epistemic uncertainty) | I/O errors (ply thickness, material, layup definition), code bugs |
| Stress-Free Temps/ Residual Curing Strain Input (COMPRO) | Many parameters can affect residual stress: local fiber volume fraction, ... | Micro-stresses are considered to be independent of meso-stresses; there are few independent measurements of residual stress. | The formulation is believed to be most accurate when the cure cycle temperature is higher than the Tg. Otherwise the residual stress calculated can be an overestimate. | Errors in material property definition, errors in coding, errors in integrating process and structural models. |
| Coupon Geometry and Load/BC Input (COMPRO, User-defined, Empirical) | Cured ply thickness variations, specimen dimensional tolerances, specimen curvatures due to residual stress/strain | | | Errors in Coupon Geometry Definition or Improper Idealization of Loading or Boundary Conditions |





Handling Uncertainty – The AIM-C Approach

- Next we must know which variables are important to us
- Complex problems have hundreds of potential uncertainties. Its time-prohibitive to spend equal effort investigating each one. We must Focus on the important “few”:
 1. Uncertainties which are likely to occur
 2. Uncertainties with a large influence on the response(s) of interest

(This evaluation is similar to simple Risk Analyses, assessing Probability of occurrence and consequences of failure)

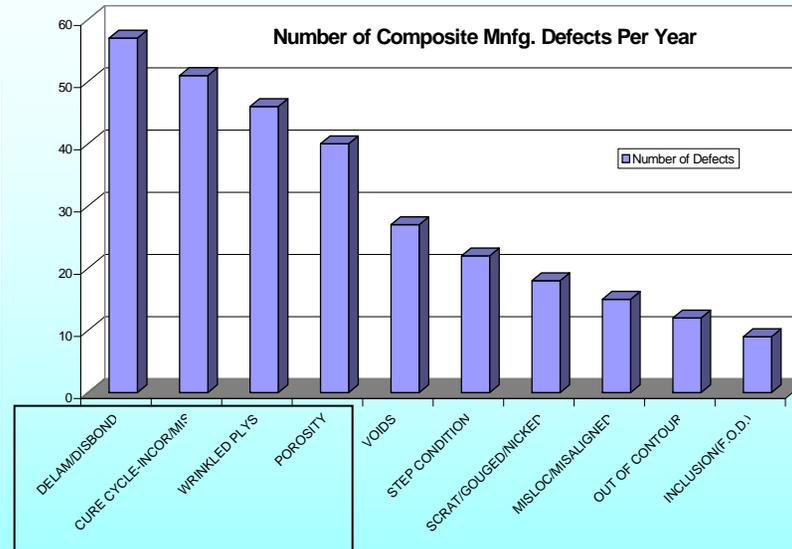


Handling Uncertainty – The AIM-C Approach

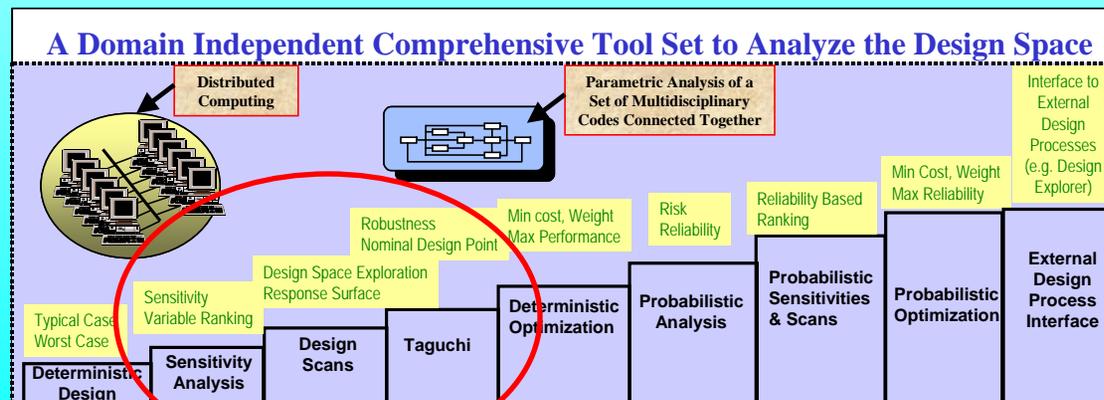
- Prior knowledge is useful in determining likelihood of occurrence.

Example: Past experience with Similar designs suggest that 3/4 of Stiffened panel defects are:

- Delaminations
- Cure Cycle Inconformities
- Ply wrinkles, or
- Voids/Porosity



- Tools such as DOE/ANOVA and Sensitivity Analysis are useful in quantifying a variable's influence on the result.



Approved for Public Release, Distribution Unlimited

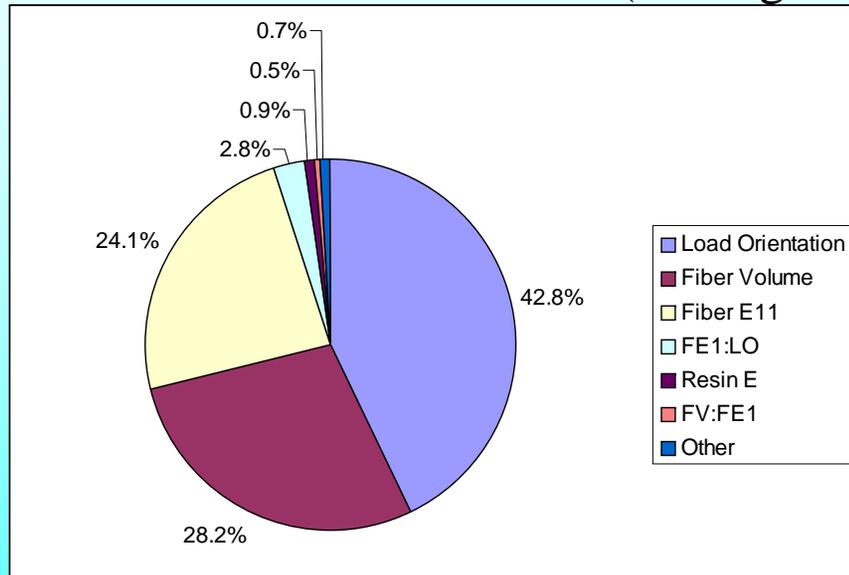




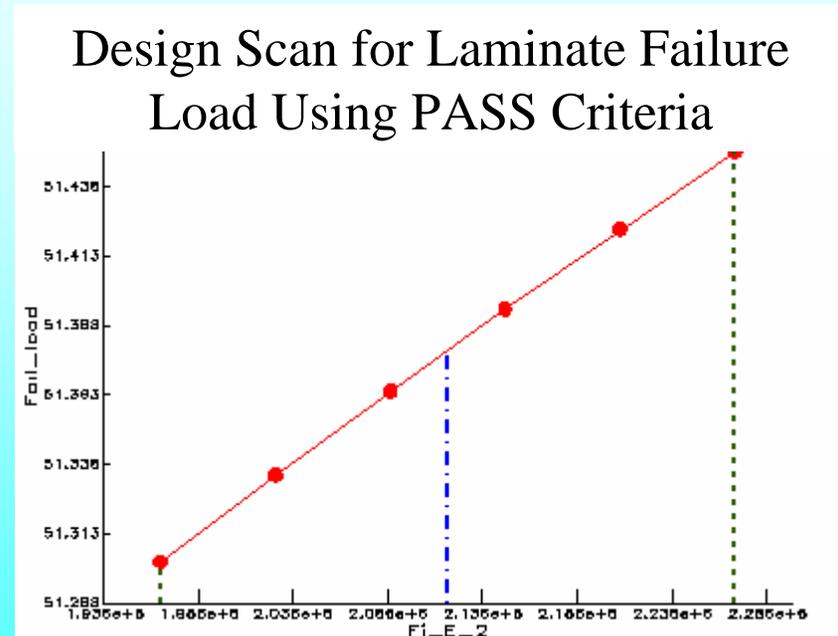
Handling Uncertainty – The AIM-C Approach



- Example – Fiber transverse modulus effect on Laminates
- Other examples:
 - Stress Free Temperature on laminates (variation was small)
 - Stiffener Pull-off (some geometric variables had little effect)



ANOVA for Laminate Axial Modulus



- As expected, Fiber Volume and Fiber E_{11} also have significant effects on laminate Modulus
- Fiber E_{22} and Resin E have very little effect (<1%)

- Transverse Fiber Modulus (E_{22}) has very little effect



Approved for Public Release, Distribution Unlimited





Handling Uncertainty – The AIM-C Approach

- Where possible, some uncertainties can be eliminated or limited by design choices. Pick the material and design to play to your strengths!
- Allows negotiation between competing response variables
 - E.g., Structural Performance and Producibility
- This is a major philosophical shift for Structures. Focus on Design Robustness rather than Absolute Mean Performance may generally yield a better “allowable” failure load.
- Tools such as DOE/ANOVA and Sensitivity Analysis are again useful.

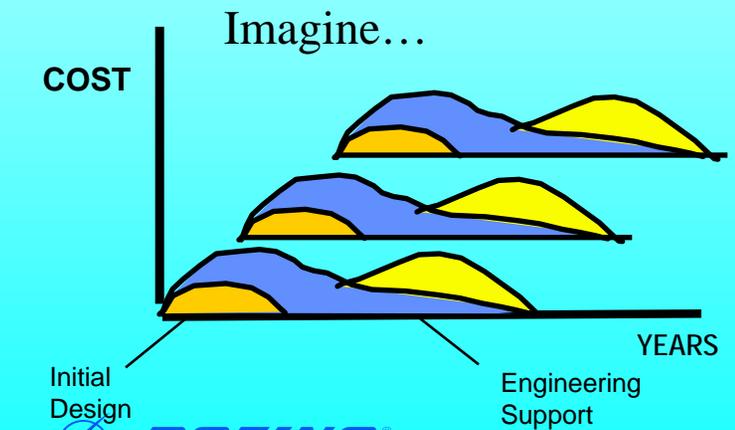
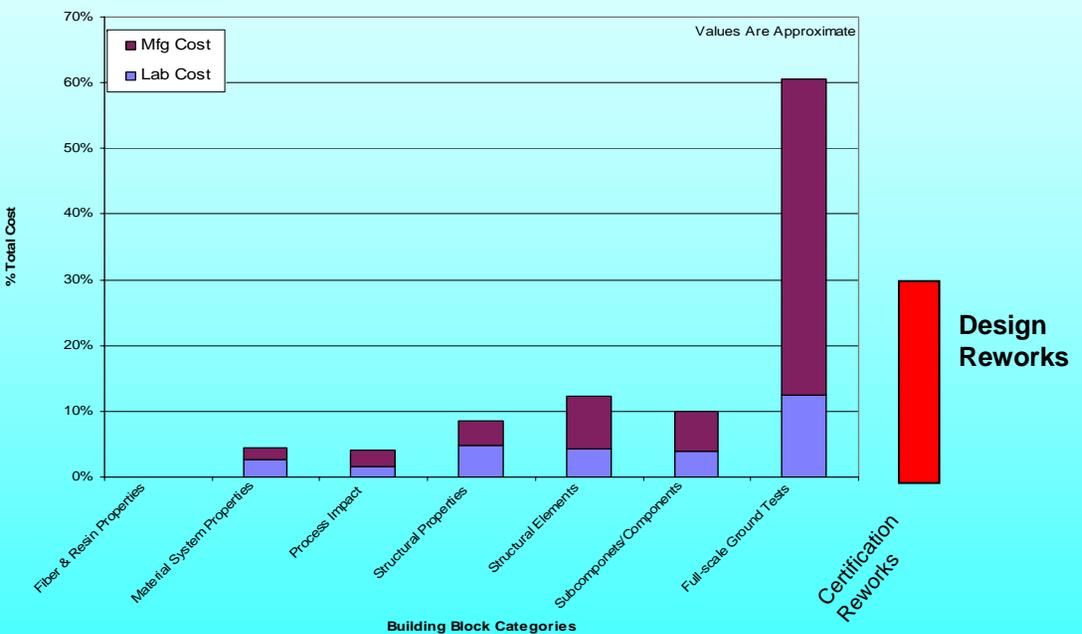
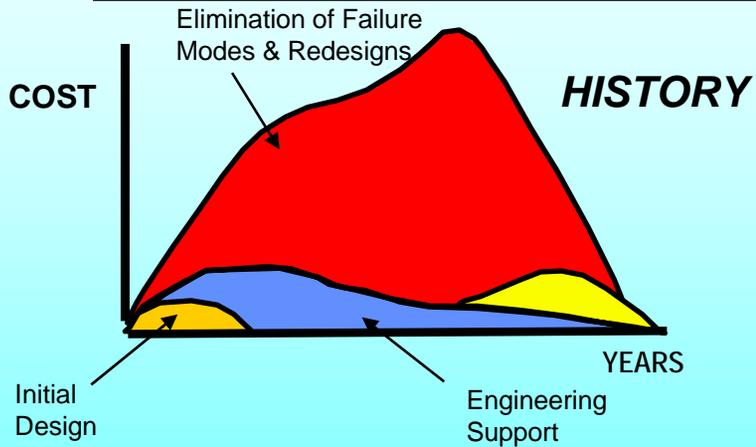


Handling Uncertainty – The AIM-C Approach

Minimizing Uncertainty by Design

Design Robustness → Avoiding Redesigns

Often, a majority of our development time and money is spent on fixing problems because we failed to choose a Robust Material, Design and/or Process the first time.



Approved for Public Release, Distribution Unlimited





Handling Uncertainty – The AIM-C Approach

Minimizing Uncertainty by Design

- Example – Hat-Stiffened Panel Design

Problem 1:

- Bondline delaminations are commonly occurring defects
- They occur at structurally-critical locations
- The failure load can be very sensitive to bondline delaminations

Problem 2:

- All dimensions have manufacturing tolerances
- In some cases, failure can be very sensitive to off-nominal dimensions

Questions:

Can we formulate a design that is much less sensitive to delaminations?

Can we minimize the effect of off-nominal dimensions on the failure load?

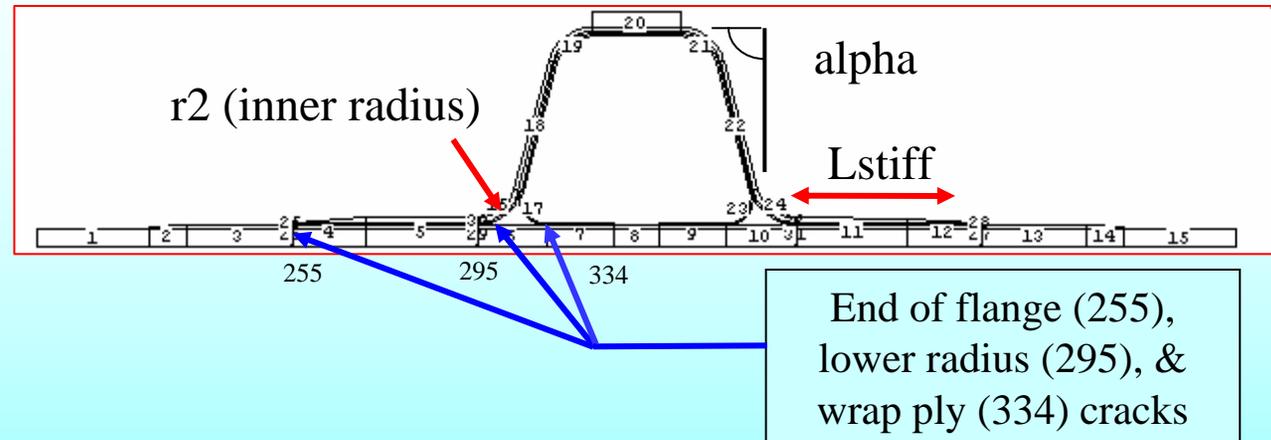


Minimizing Uncertainty by Design

HSP Pull-off Modeling with Defects

- Utilizes the following modules within AIM-C

- Fiber
- Resin
- Lamina
- Fabric



- Allows for both material and geometric variability
 - Focused on length of stiffener, leg angle, and lower radius
- RDCS aided in the following tasks:
 - Determining the sensitivity of embedded flaws to geometric parameters
 - Pinpointing optimal geometric parameters that minimize SERR in
 - Lower radius (multiple delamination locations at “nugget”)
 - Length of horizontal stiffener section



Minimizing Uncertainty by Design

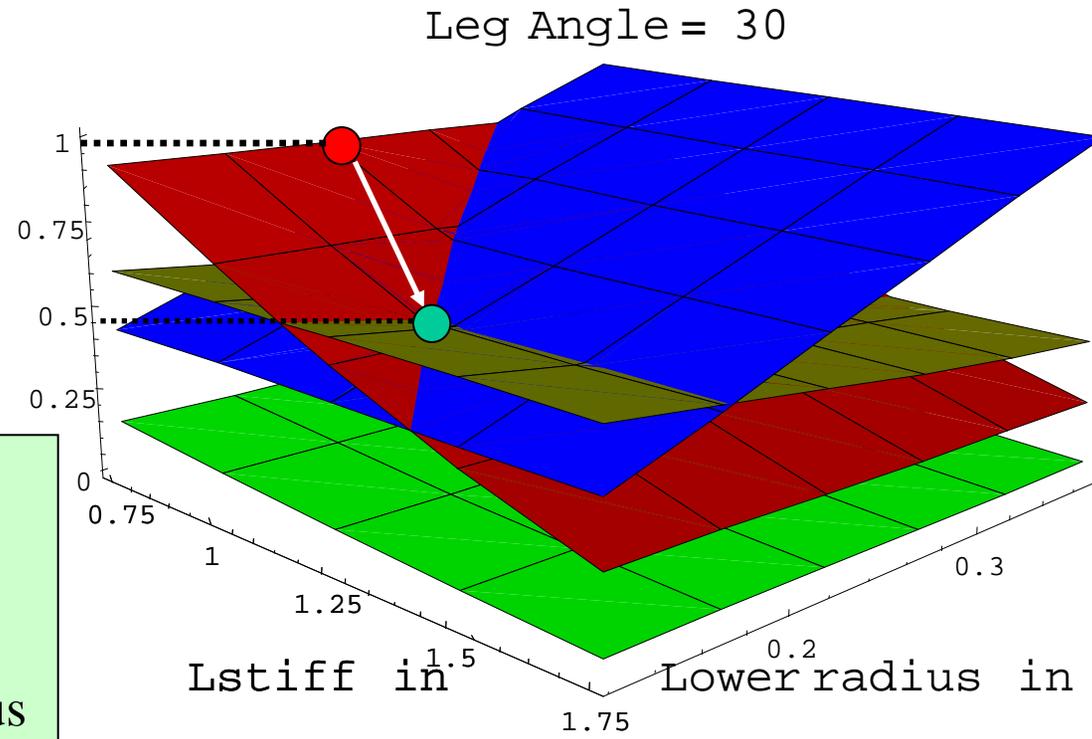
Geometry Effect on Radius and Edge-of-Flange Delaminations

SERR Data, Leg Angle = 30 degrees

Large L_{stiff} and smaller lower radius minimizes energy available to propagate a crack

SERR in-lbs in²

- Red: G_I EOF
- Blue: G_I Lower Radius
- Green: $G_{II}/4$ EOF
- Brown: $G_{II}/4$ Lower Radius



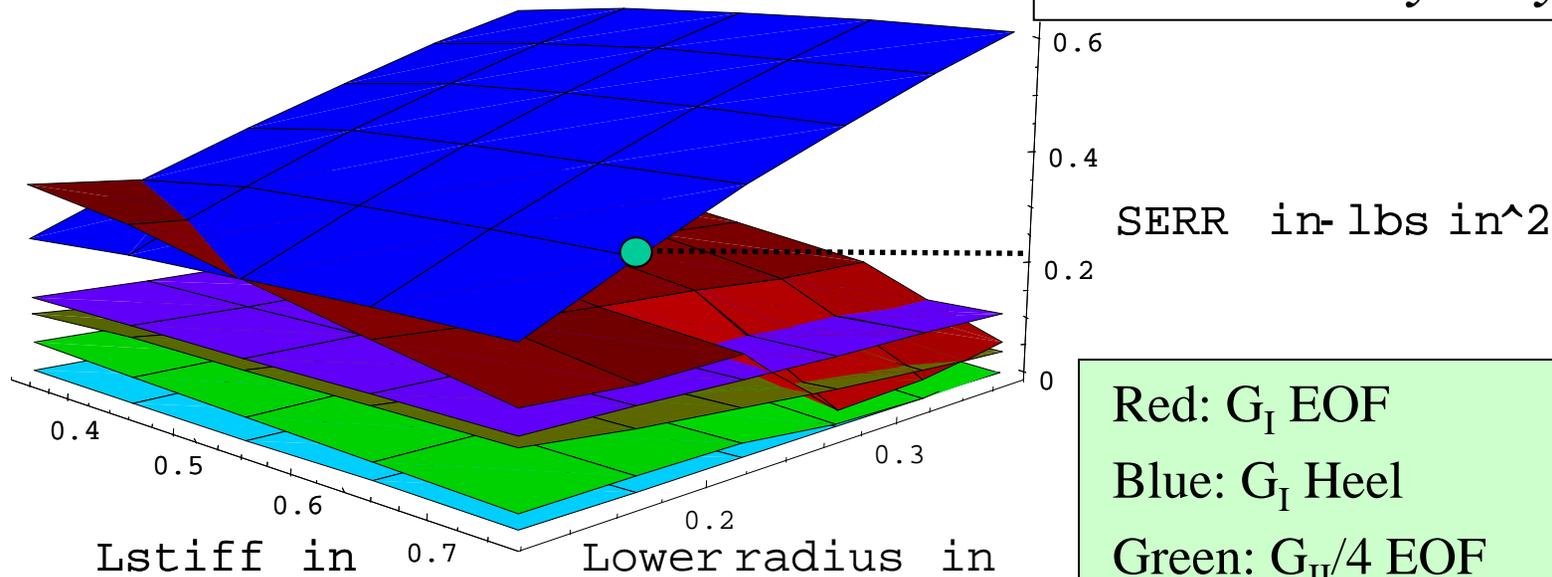
A local SUBLAM model run suggests that the lower radius and stiffer leg angle could be designed to minimize the effect of potential radius delaminations. Sufficient stiffer Leg length controls Edge-of-Flange Delamination



Minimizing Uncertainty by Design

SERR – Short Bay Width

Leg Angle = 20



Adding wrap and shortening bay width reduces SERR magnitude trends similar to Wide Bay study.

- Red: G_I EOF
- Blue: G_I Heel
- Green: $G_{II}/4$ EOF
- Brown: $G_{II}/4$ Heel
- Light Blue: G_I Wrap/Plank
- Purple: $G_{II}/4$ Wrap/Plank

With reduced bay width, wrap plies, and leg angle of 20°, G_{II} is no longer critical and SERRs are all generally reduced.

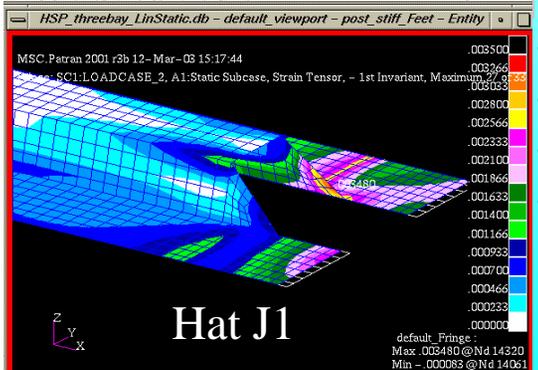
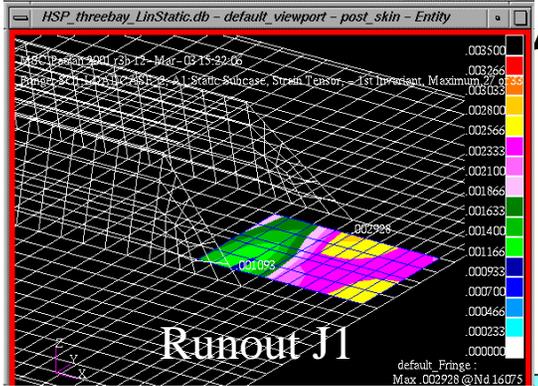
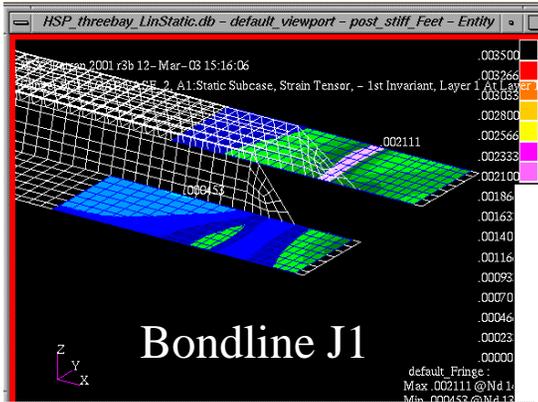
Approved for Public Release, Distribution Unlimited



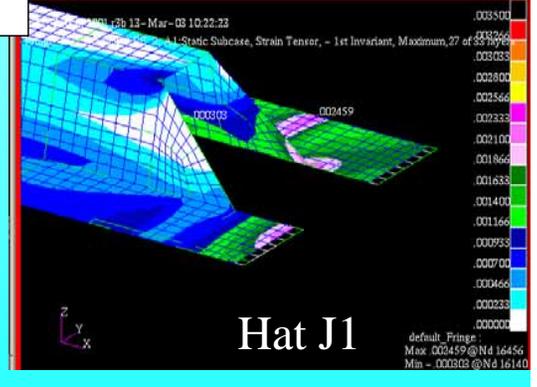
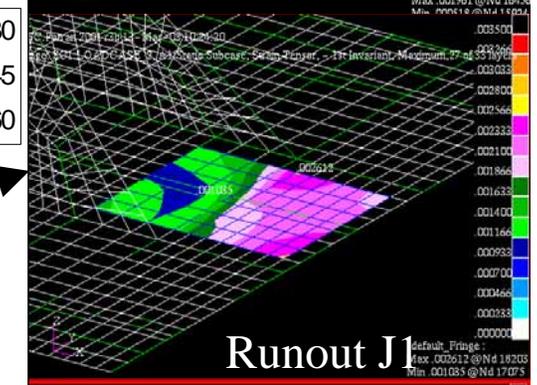
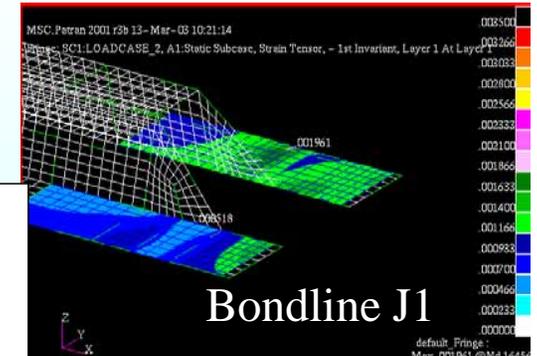
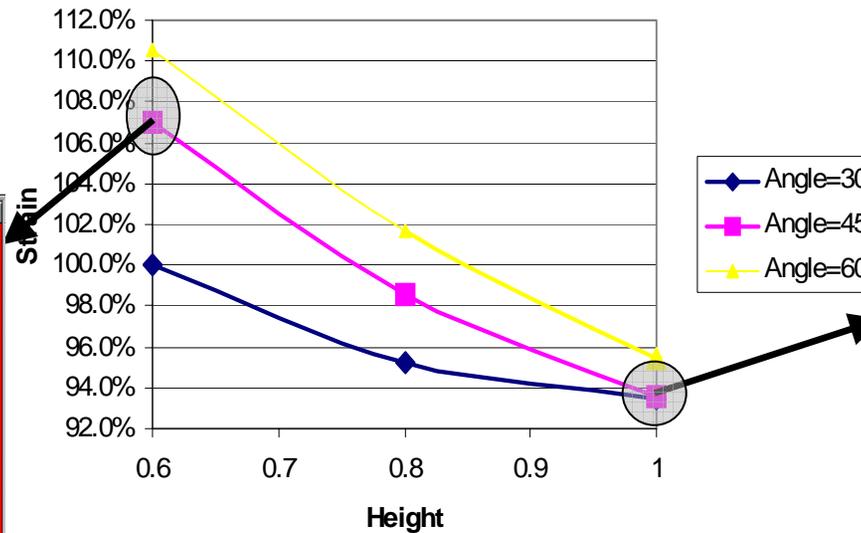


Minimizing Uncertainty by Design

Hat Height, Width, and Runout Angle Study



Critical Load Case: 2



- Tall hats have lower J_1 in all regions
- Shallow runout angle is better
- Tall hats are less sensitive to runout angle
- Shallow runout angles make design less sensitive to hat height



Approved for Public Release, Distribution Unlimited



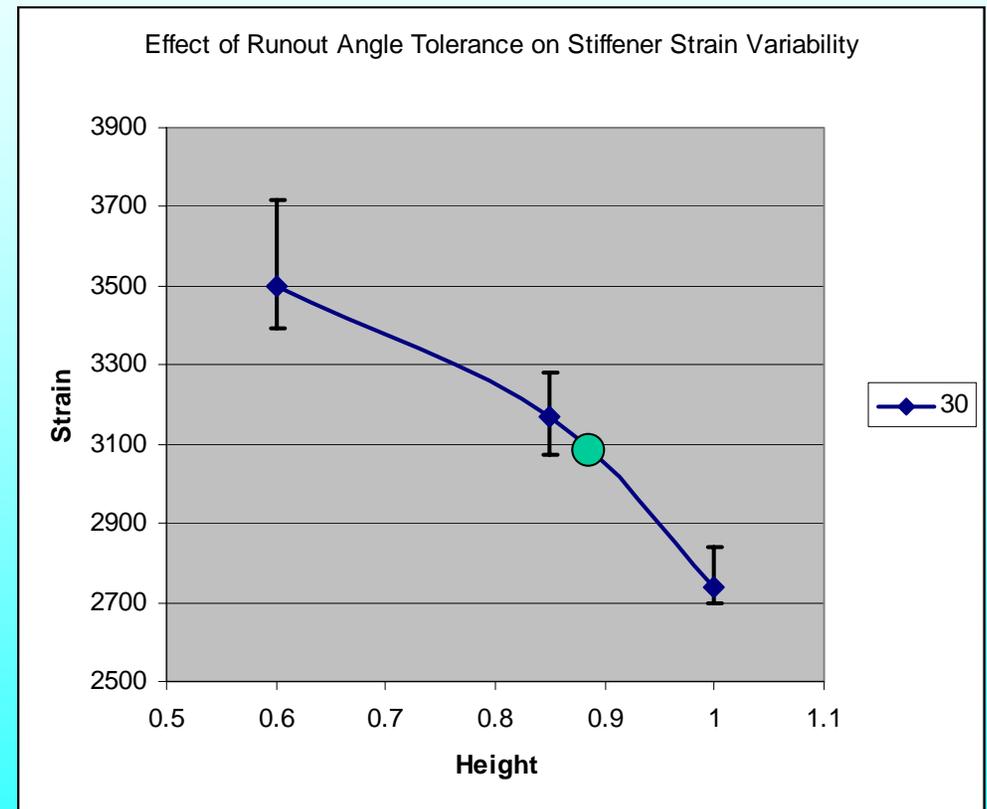
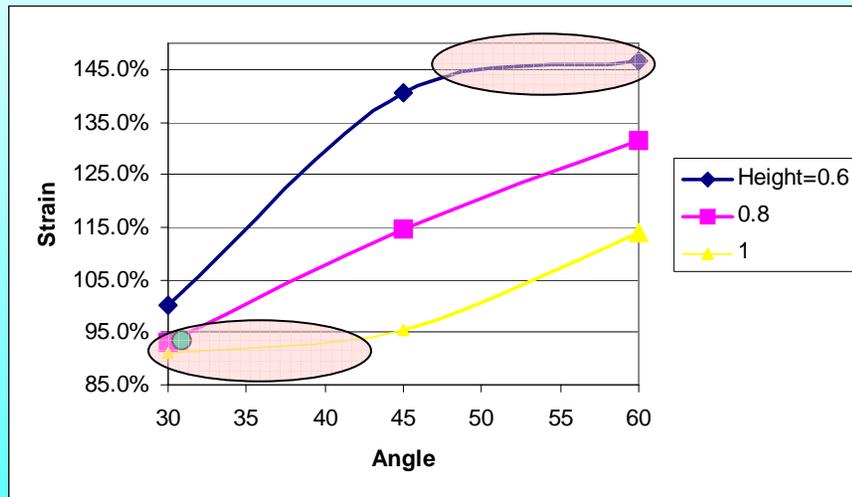


Minimizing Uncertainty by Design

Robustness to Geometric Variability – Hat Height, Width, Runout

Design on the Flat!

Tall Hats with shallow runouts or short hats with steep runouts are less sensitive to runout angle tolerance

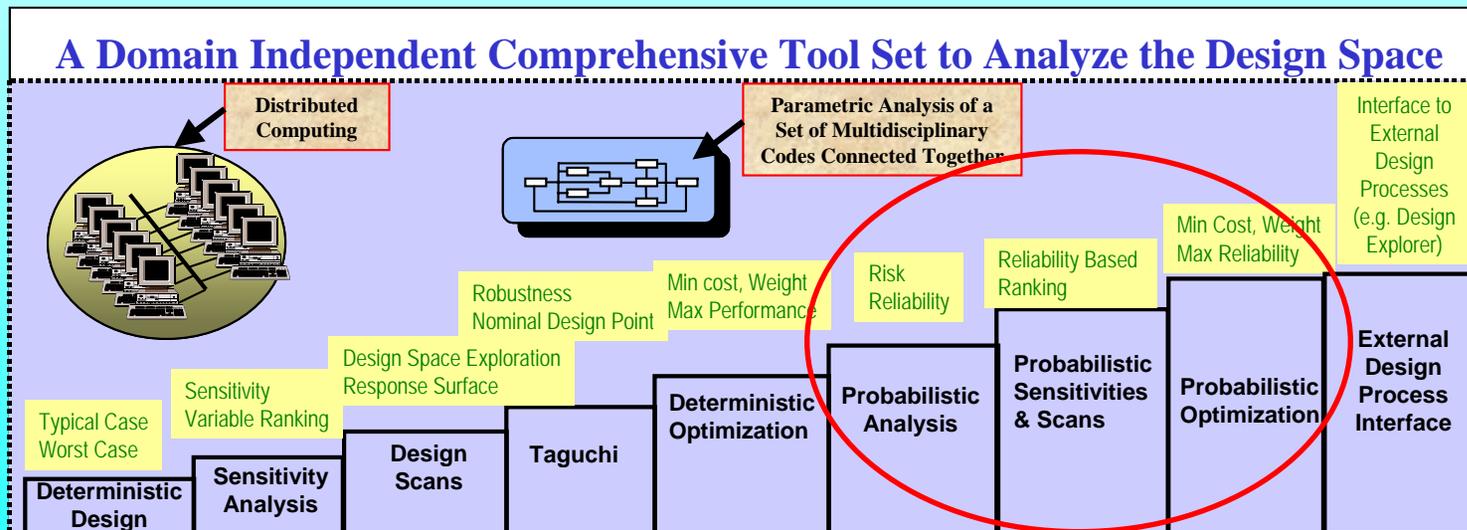




Handling Uncertainty – The AIM-C Approach

Quantifying Uncertainty

- If its important, and you can't design it out, quantify it.
- Another change from current philosophy. Currently only done with coupon allowables. Other variation is considered covered in “material scatter”, covered by factors, or worst-case assumptions.
- Testing or Probabilistic Analysis Tools are applied.





Handling Uncertainty – The AIM-C Approach

Quantifying Uncertainty

Recent Enhancements

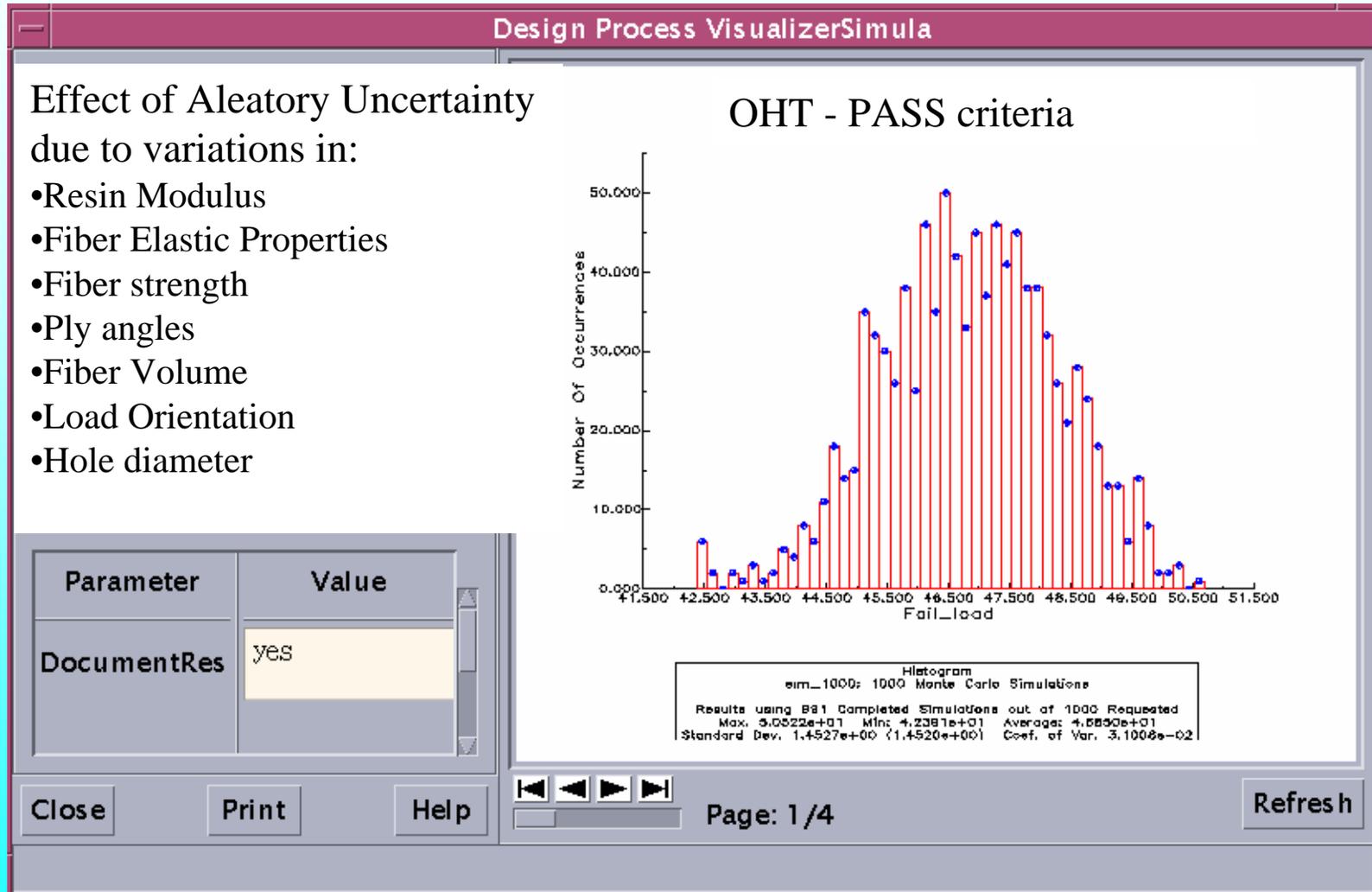
- Greatly expanded the operating space of uncertainty analysis
 - Continuous, discrete and enumerated variable types
 - Sensitivity analysis on mixed space and constrained design space exploration
- Integration of external uncertainty analysis plug-ins with RDCS
 - Advanced design of experiments – Design Explorer
- Probabilistic (Robust) Optimization
 - A capability to define statistical parameters as design variables
 - One of the tools for use in the current on-going model-test integration task



Handling Uncertainty – The AIM-C Approach

Quantifying Uncertainty

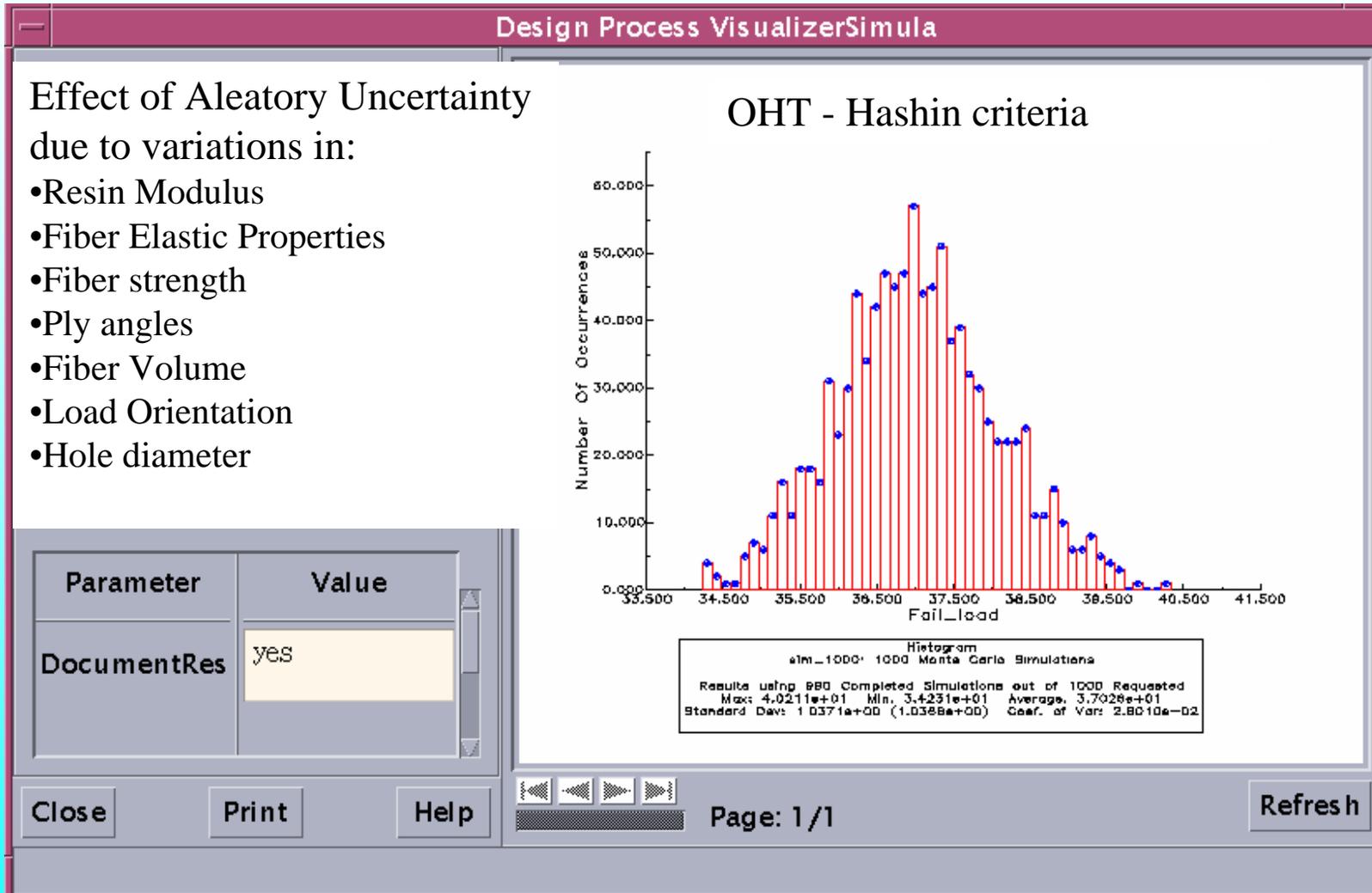
- Example – OHT Laminate Monte-Carlo Simulation





Handling Uncertainty – The AIM-C Approach

- Example – OHT Laminate Monte-Carlo Simulation

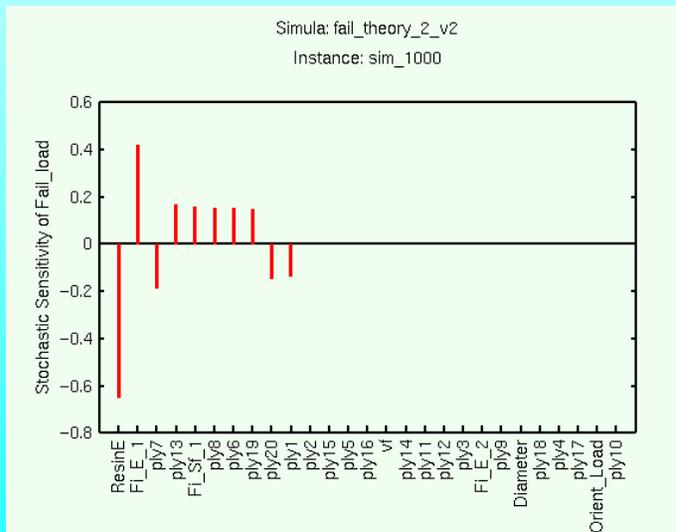




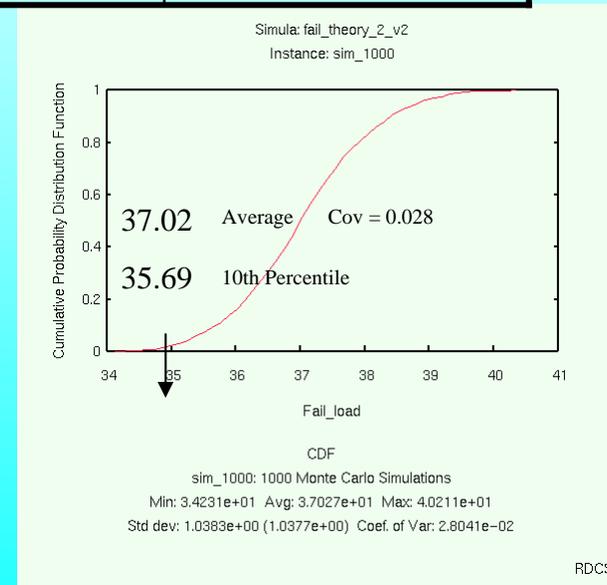
Handling Uncertainty – The AIM-C Approach

- Example – OHT Laminate Monte-Carlo Simulation

| | Test | Hashin | Phase Avg. |
|--------------------------|--------|--------|------------|
| Mean | 37.274 | 34.231 | 42.39 |
| Std.Deviation | 1.683 | 1.0371 | 1.4527 |
| Coefficient of Variation | .04517 | .02801 | .031 |



**10/80/10
Layup**





Handling Uncertainty – The AIM-C Approach

Quantifying Uncertainty

HSP Robustness to Flaws, Geometric and Material Variability

Probabilistic Analysis – Monte Carlo Simulation

Problem Definition: Random variables listed below, all other parameters same as Study 09.

Geometry *only*

- Length of stiffener flange
(Mean = 1.25”, SD = 0.015”)
- Leg angle (Mean = 20°, SD = 1.5°)
- Lower radius
(Mean = 0.2”, SD = 0.015”)

Geometry + Material

- Length of stiffener (SD = 0.015”)
- Leg angle (SD = 0.015”)
- Lower radius (SD = 1.5°)
- Fiber volume (5% COV)
- Fiber modulus (5% COV)
- Resin modulus (5% COV)

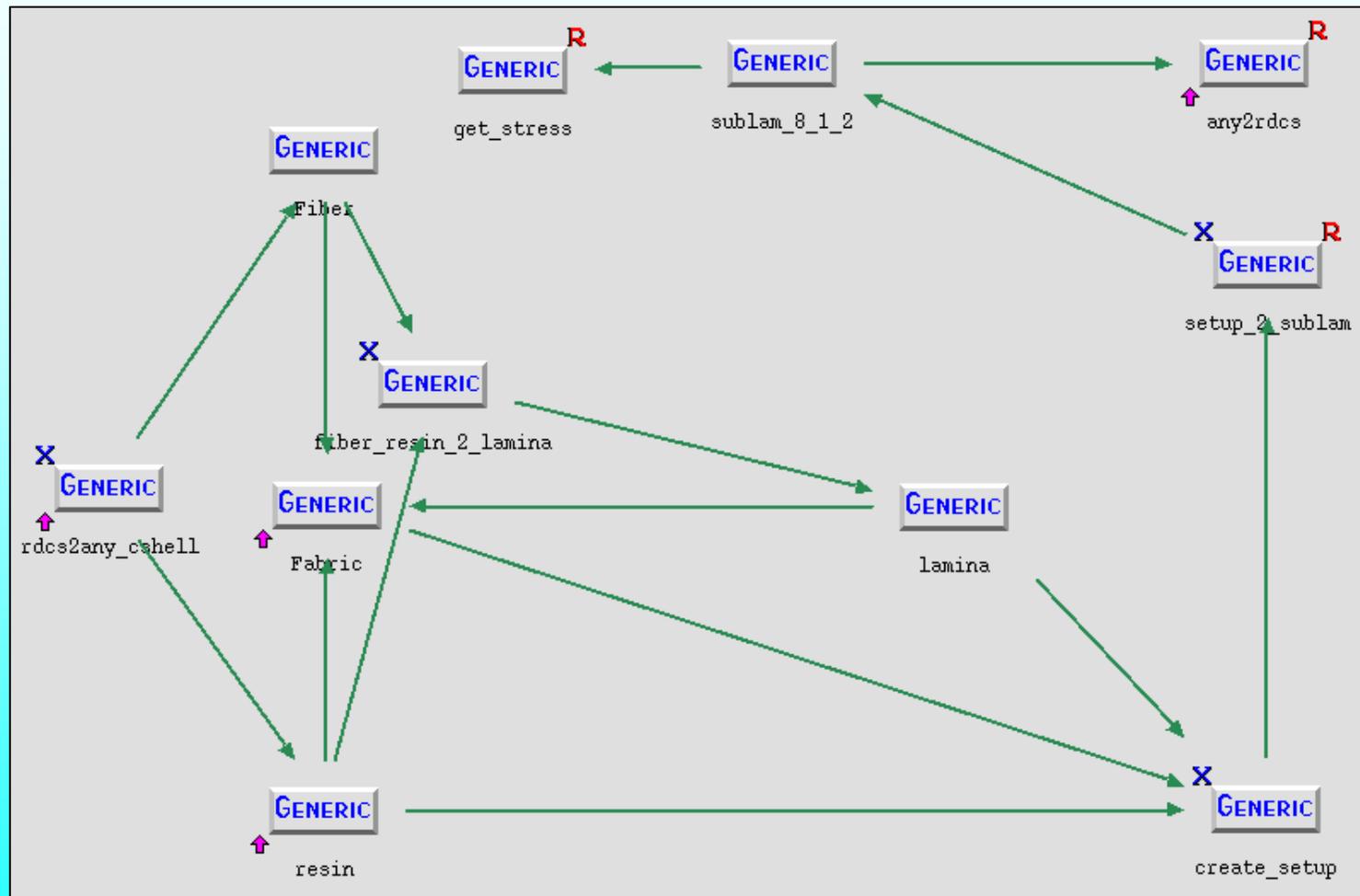
Normal distributions were used for all input parameters,



Quantifying Uncertainty

Robustness to Flaws, Geometric and Material Variability

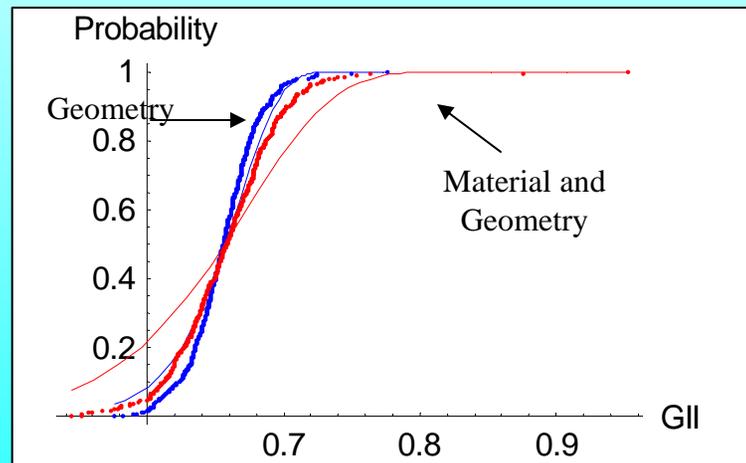
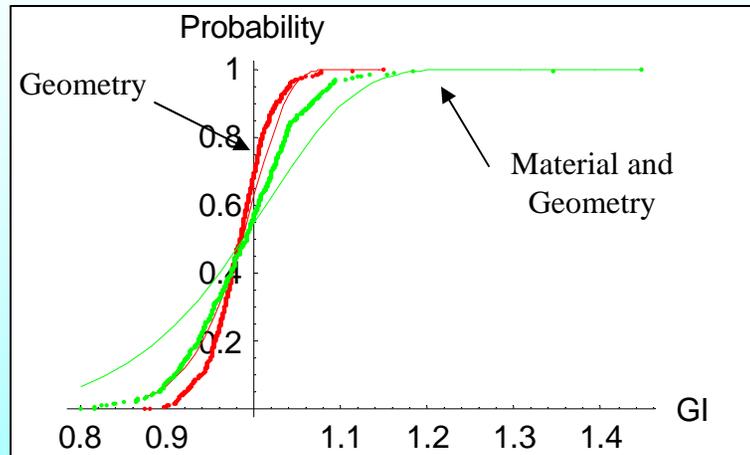
Probabilistic Analysis – RDCS Math Model





Quantifying Uncertainty

Robustness to Flaws, Geometric and Material Variability RDCS Results



- Numerical values reported for 90 lb/in pull off load
- Mode I and II SERR at end of flange drives failure results
- Variations in crack driving force increase significantly when variability in material elastic constants are added:

$$\bullet SD_{GI} \quad 0.036 \rightarrow 0.068$$

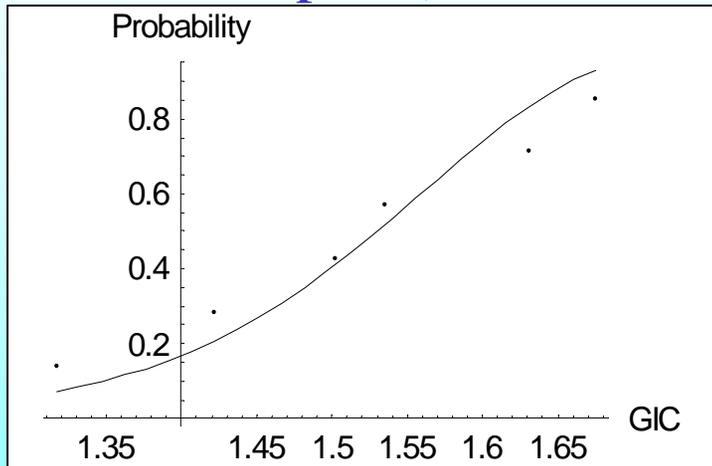
$$\bullet SD_{GII} \quad 0.026 \rightarrow 0.041$$



Quantifying Uncertainty

Robustness to Flaws, Geometric and Material Variability

Variation in Critical Failure Properties by Test Coupon (DCB and ENF) Experimental Results

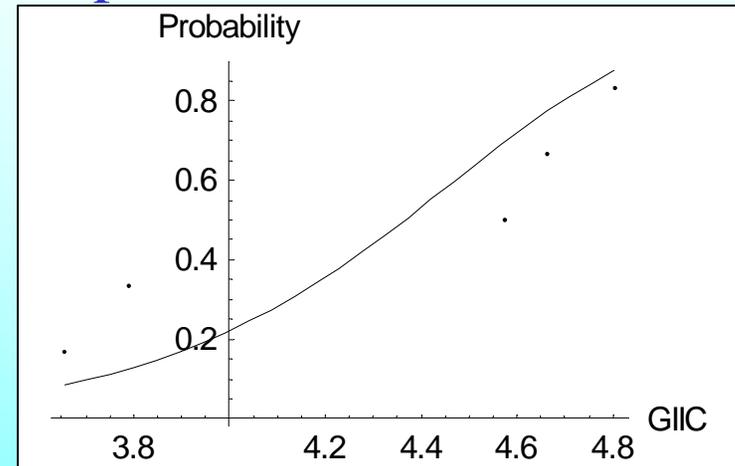


Mean = 1.51

St. Dev. = 0.132

No. of Specimens = 6

Weibull B-Value = 1.0



Mean = 4.30

St. Dev. = 0.533

No. of Specimens = 5

Weibull B-Value = 2.24

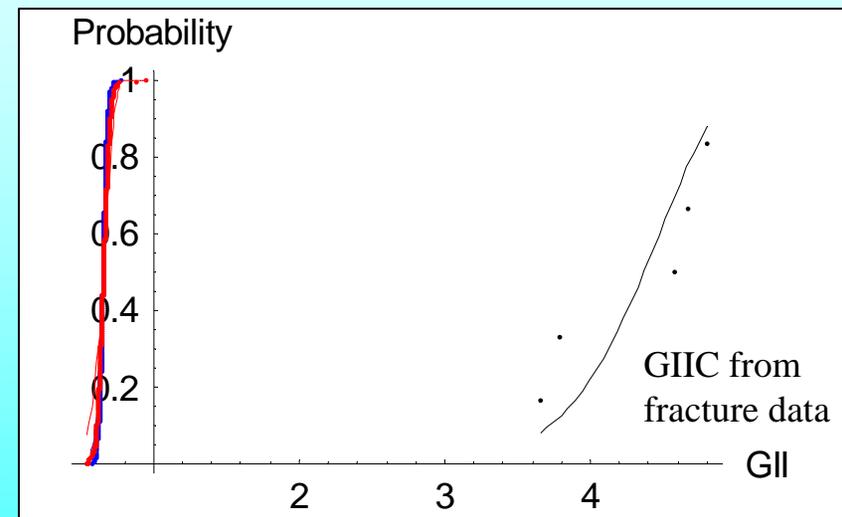
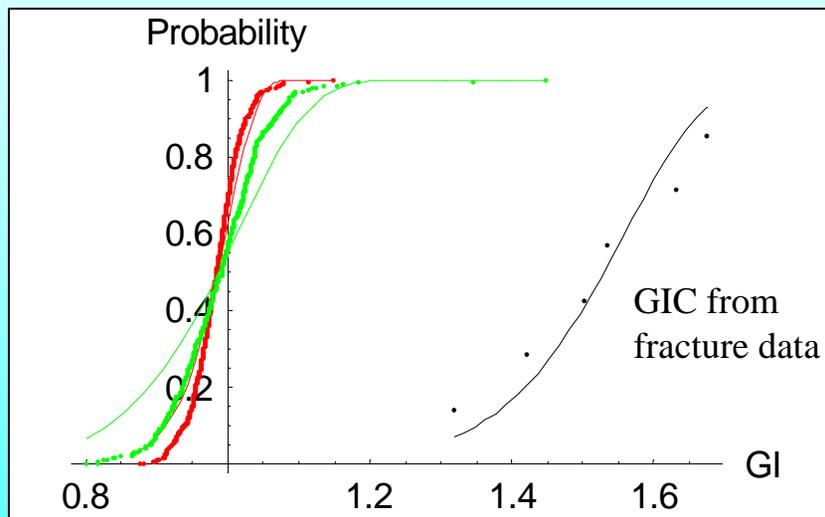


Quantifying Uncertainty

Robustness to Flaws, Geometric and Material Variability

Comparison of Variabilities

CDFs of SUBLAM SERRs and critical SERRs
from experimental fracture data



Note: Materials measured resistance to crack growth (Critical SERR) is MUCH more variable than computed variations in crack driving force due to other Material/Geometry variation