



# Implementing Digital Terrain Data In KA-STAP

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# Agenda

- Introduction/Background
- Approach
- 200 meter NLCD (national land cover data) data – injected targets
- 30 meter NLCD data – moving target simulator (MTS) targets
- Seeing what the sensor sees
- 10 meter DEM data
- Agile Intelligent Radar System (AIRS)
- Summary/Future Plans



# Introduction/Background



- Can STAP performance be improved by choosing secondary data based upon a priori map data?
- Cell Averaging Symmetric Algorithm - choose secondary data plus and minus  $N/2$  range rings from the test ring (omitting guard cells)
  - Implicit assumption - nearby range rings of the test ring are homogeneous with respect to terrain and are representative of the test ring.

Conjectures:

- Case I – Homogeneous Terrain Environment – Expect equal performance
- Case II – Heterogeneous Terrain Environment – Expect improved performance



# Objective/Proposed Algorithm



- Single-Bin Post-Doppler STAP
- Basic Assumption - Major clutter competing with the target cell is due to the patch of Earth within the same test ring that passes through the same Doppler filter
- Picking secondary range-Doppler cells that have the “same” terrain as the test cell will provide better performance than the cell averaging symmetric algorithm.



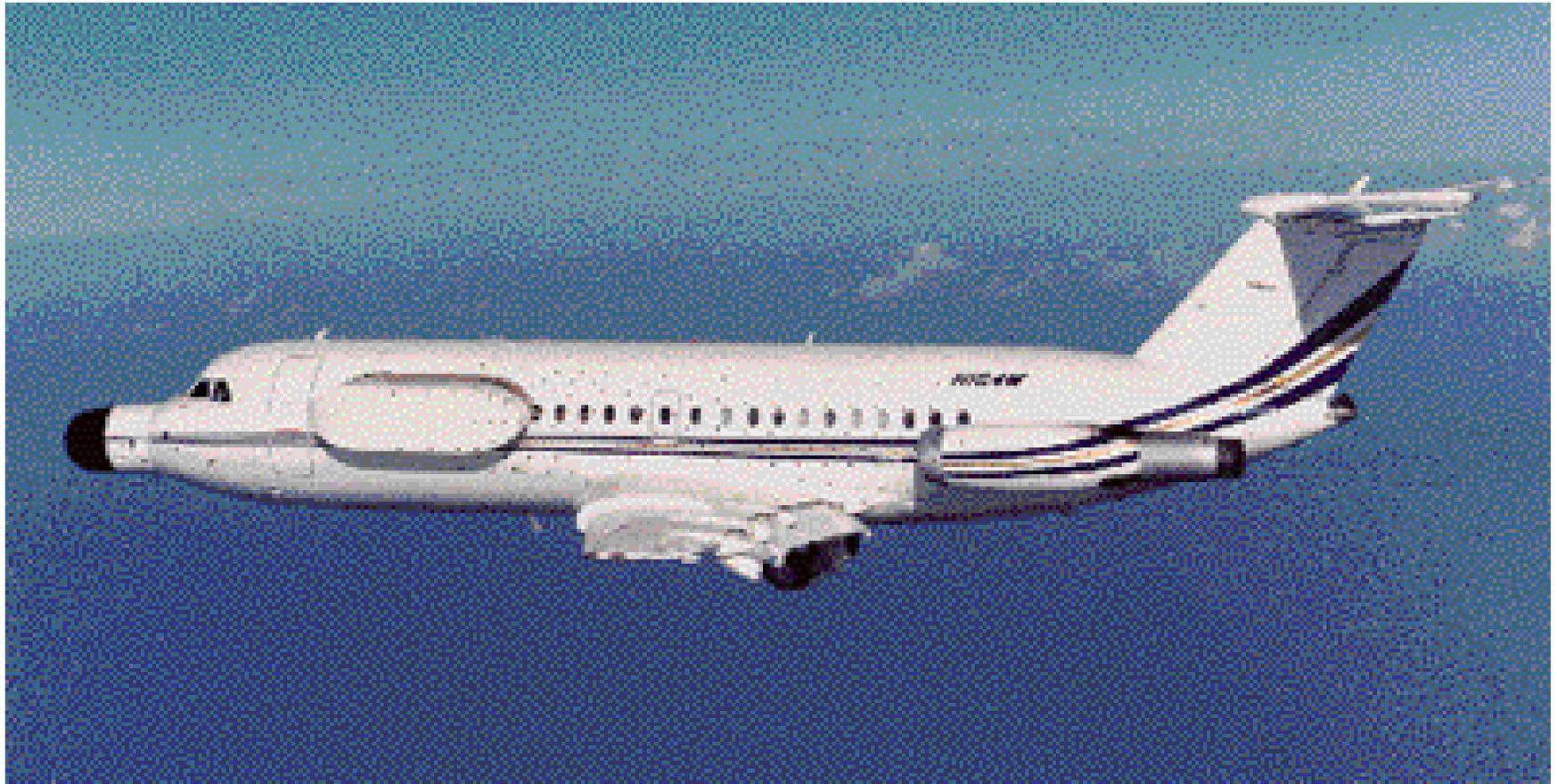
# Multi-channel Airborne Radar Measurement (MCARM) Program



- USAF RL/SN Measurement program – mid 1990's
- Side looking L-Band radar
- 2 by 11 Channel linear array including sum and delta analog beamformers
- 120 Meter resolution with approximate 500 range bins
- 128 Pulses within a coherent processing interval (CPI)
- Returns were typically unambiguous in Doppler

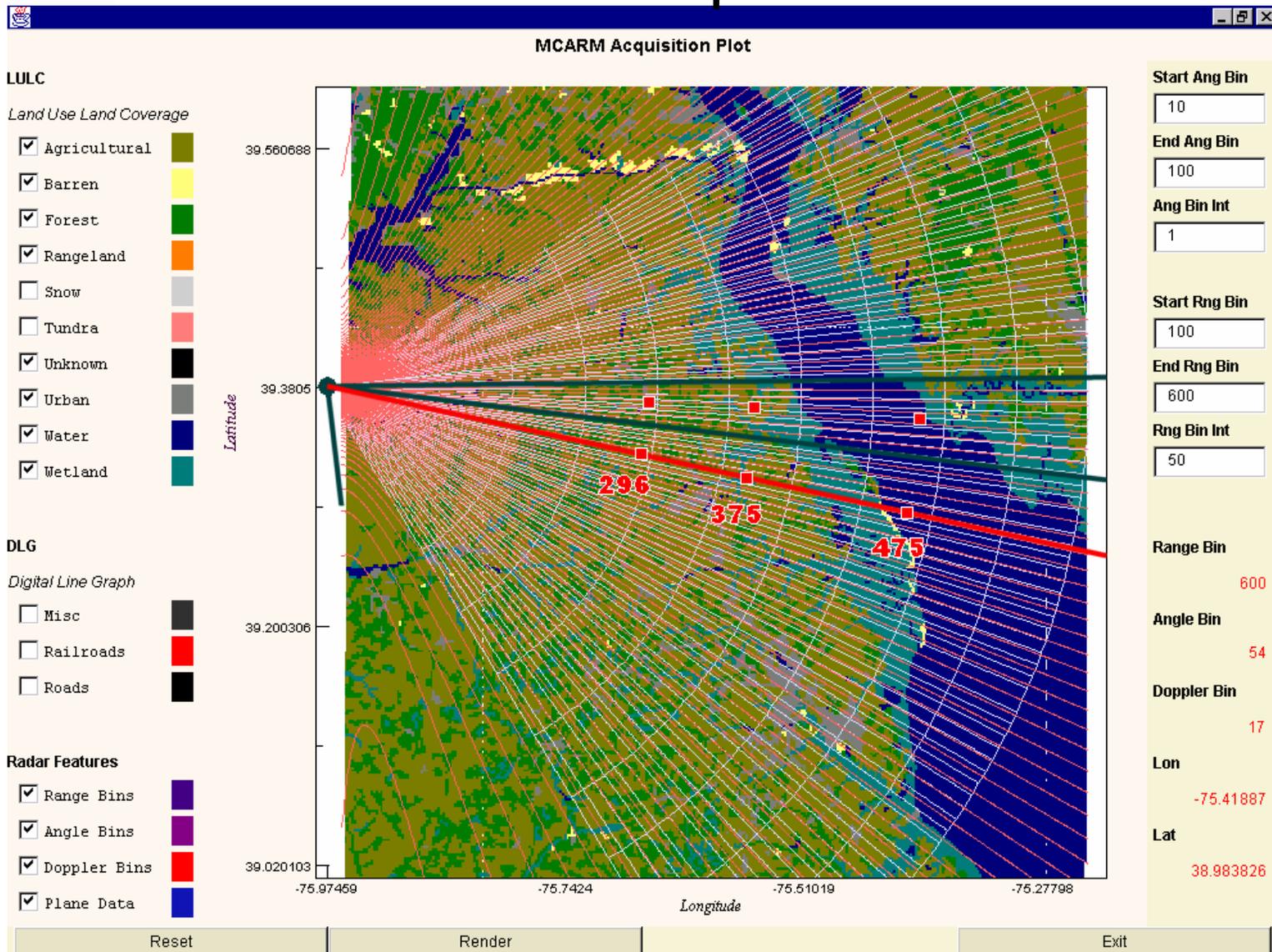


# Bac 1-11





# MCARM Radar Registration With Map Data



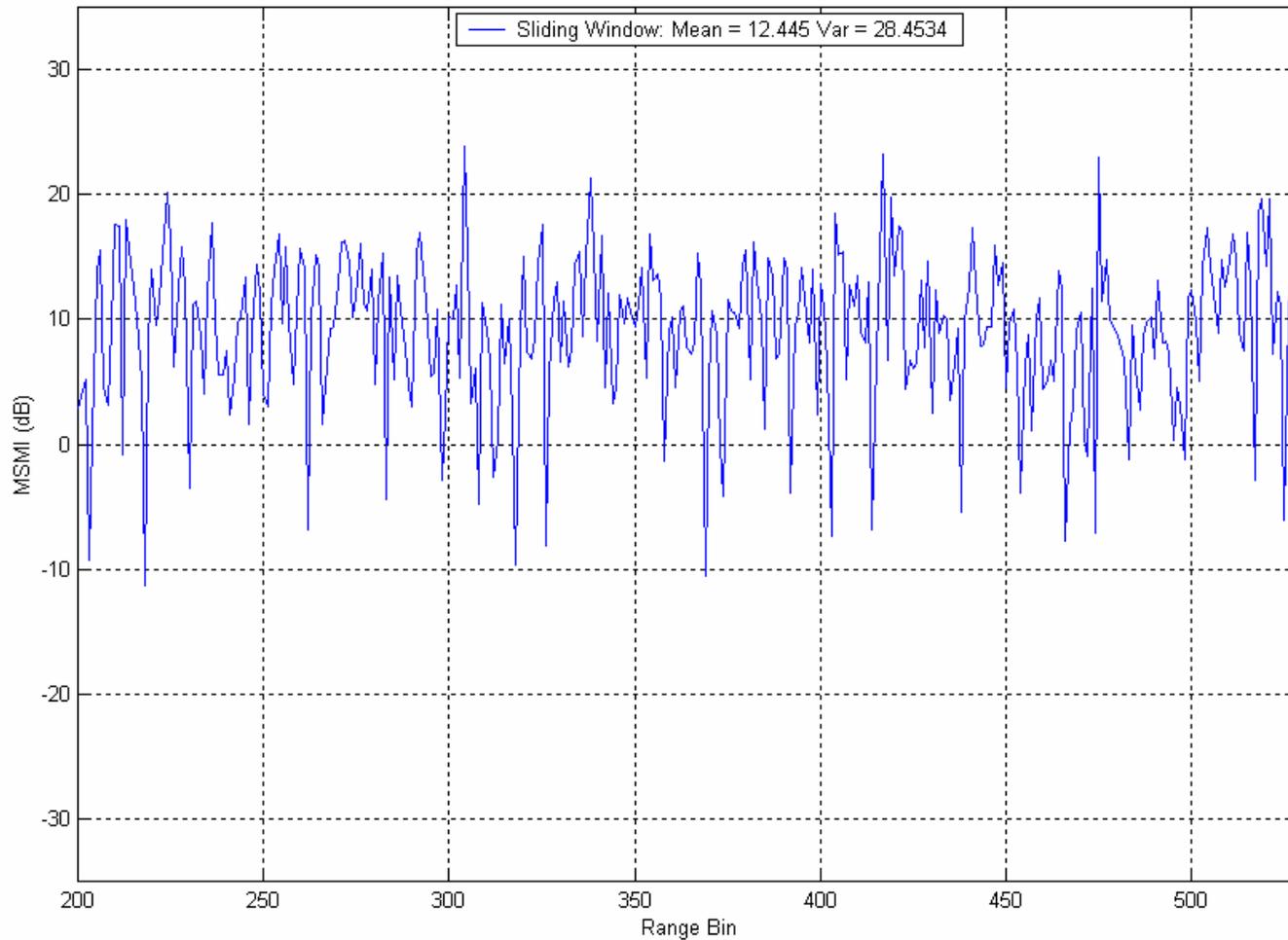


# Terrain Data Selection

- 21 terrain classifications were used, 200 meter (target injection) and 30 meter (MTS targets) resolution – National Land Cover Data (NLCD)
- Delmarva Peninsula is relatively flat – No elevation or digital line graph data were used
- Created a terrain vector from the 21 classification codes for each range-Doppler cell
- Accounted for differences in number of NLCD patches per range-Doppler cell
- Developed an algorithm to choose “like” range-Doppler cells by using a Euclidian distance measure between the normalized terrain vectors

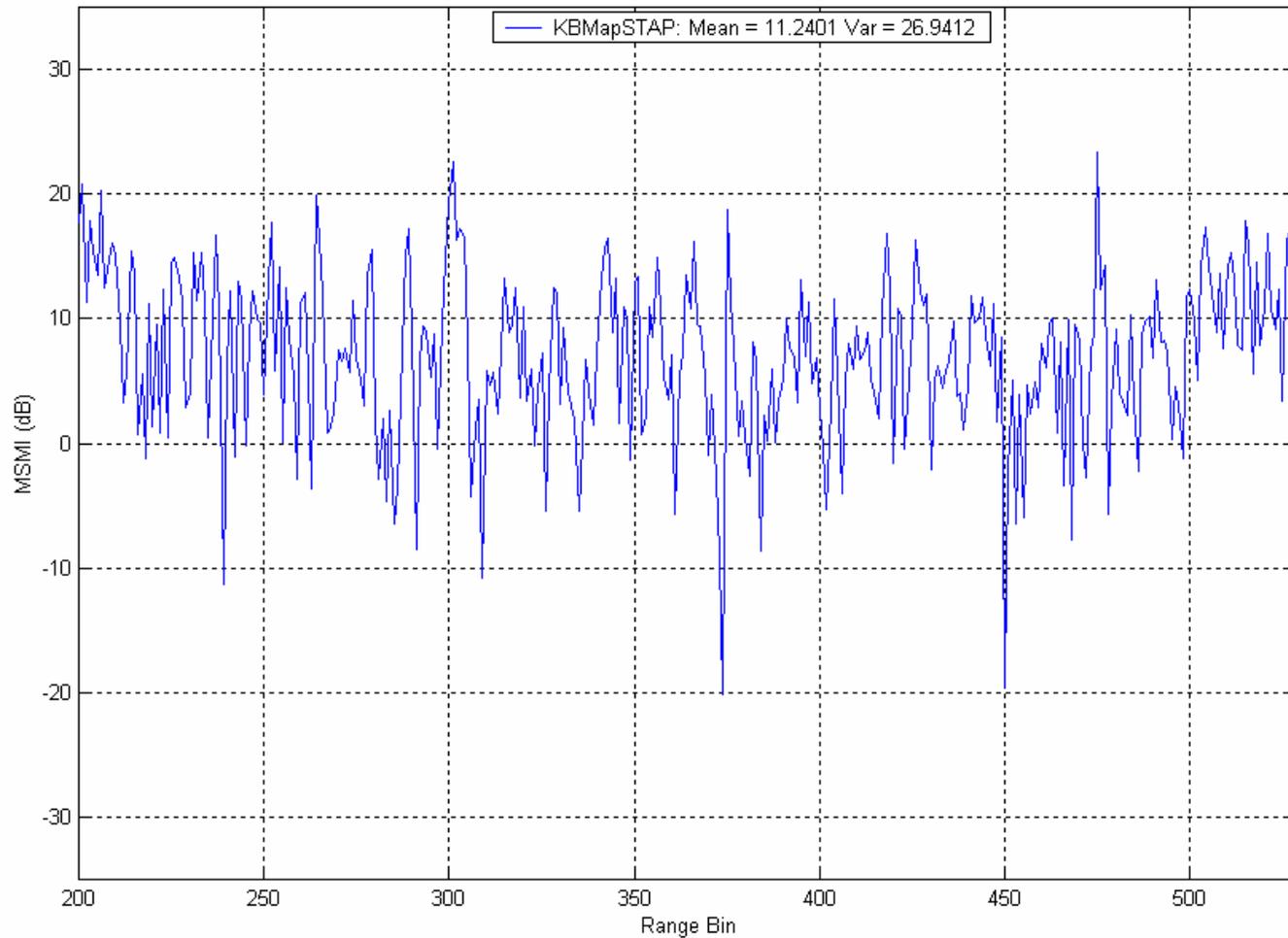


# MSMI Output Using Sliding Window Algorithm With Injected Target At Range Bin 475



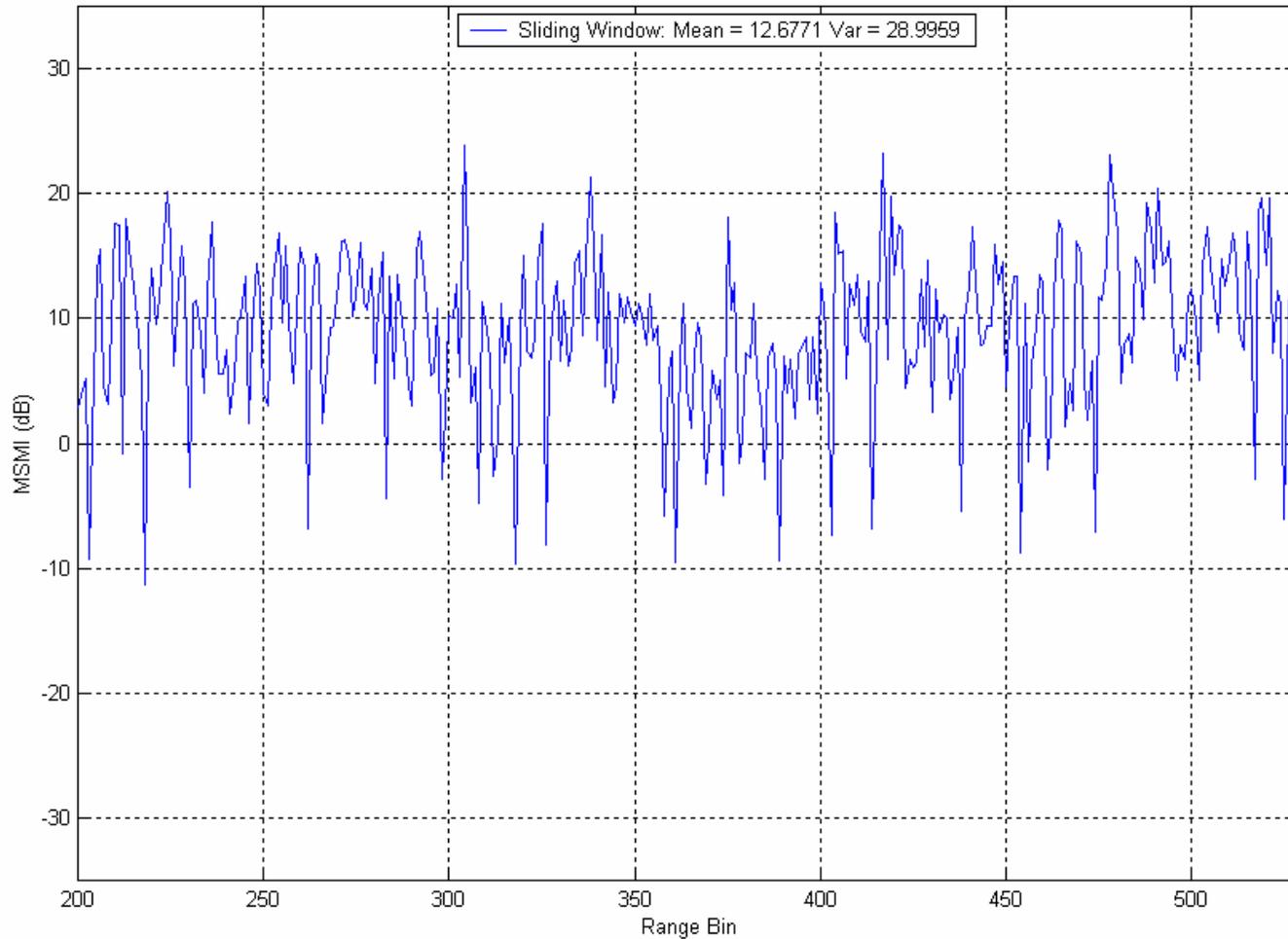


# MSMI Output Using KMapSTAP Algorithm With Injected Target At Range Bin 475



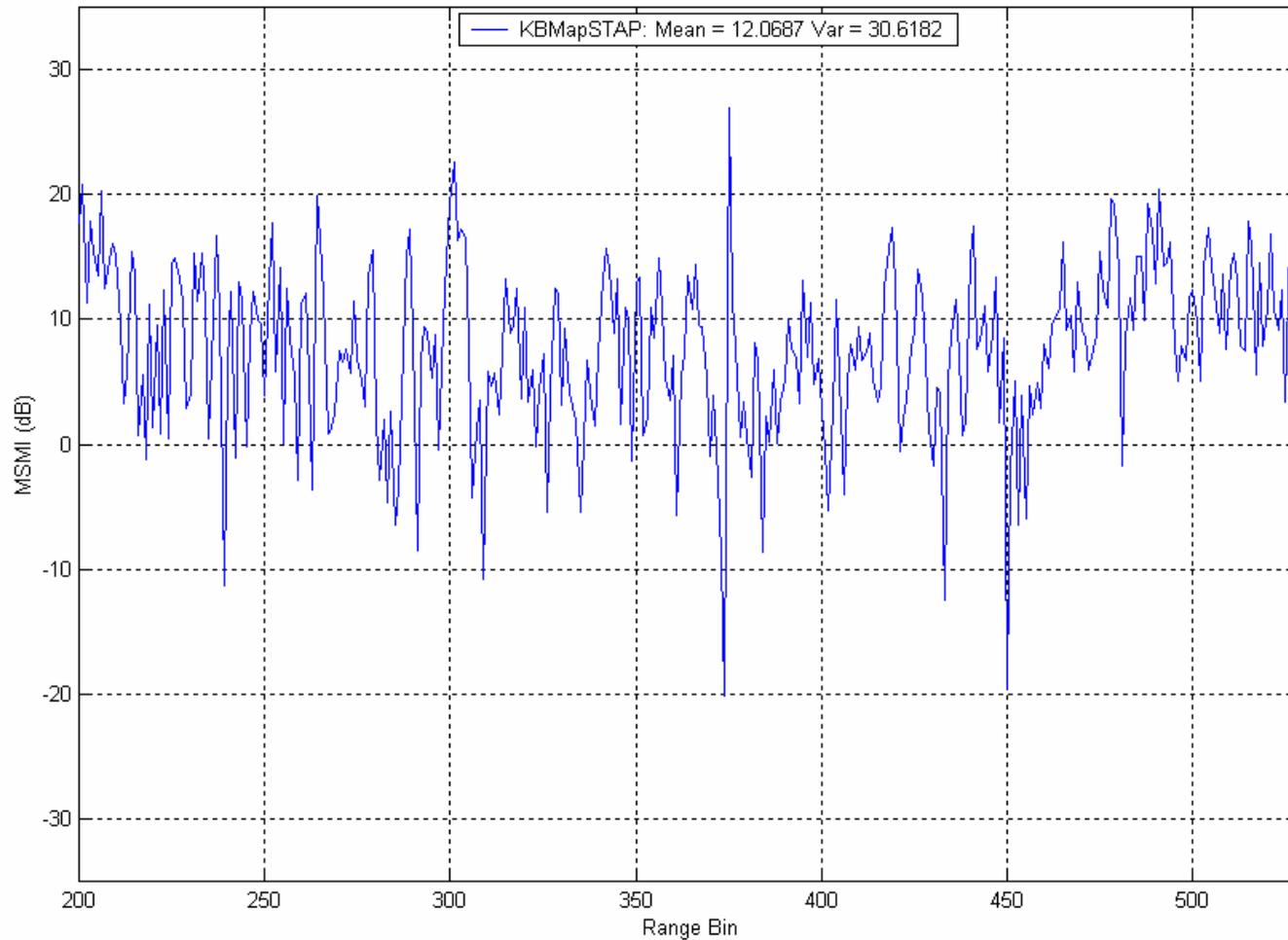


# MSMI Output Using Sliding Window Algorithm With Injected Target At Range Bin 375



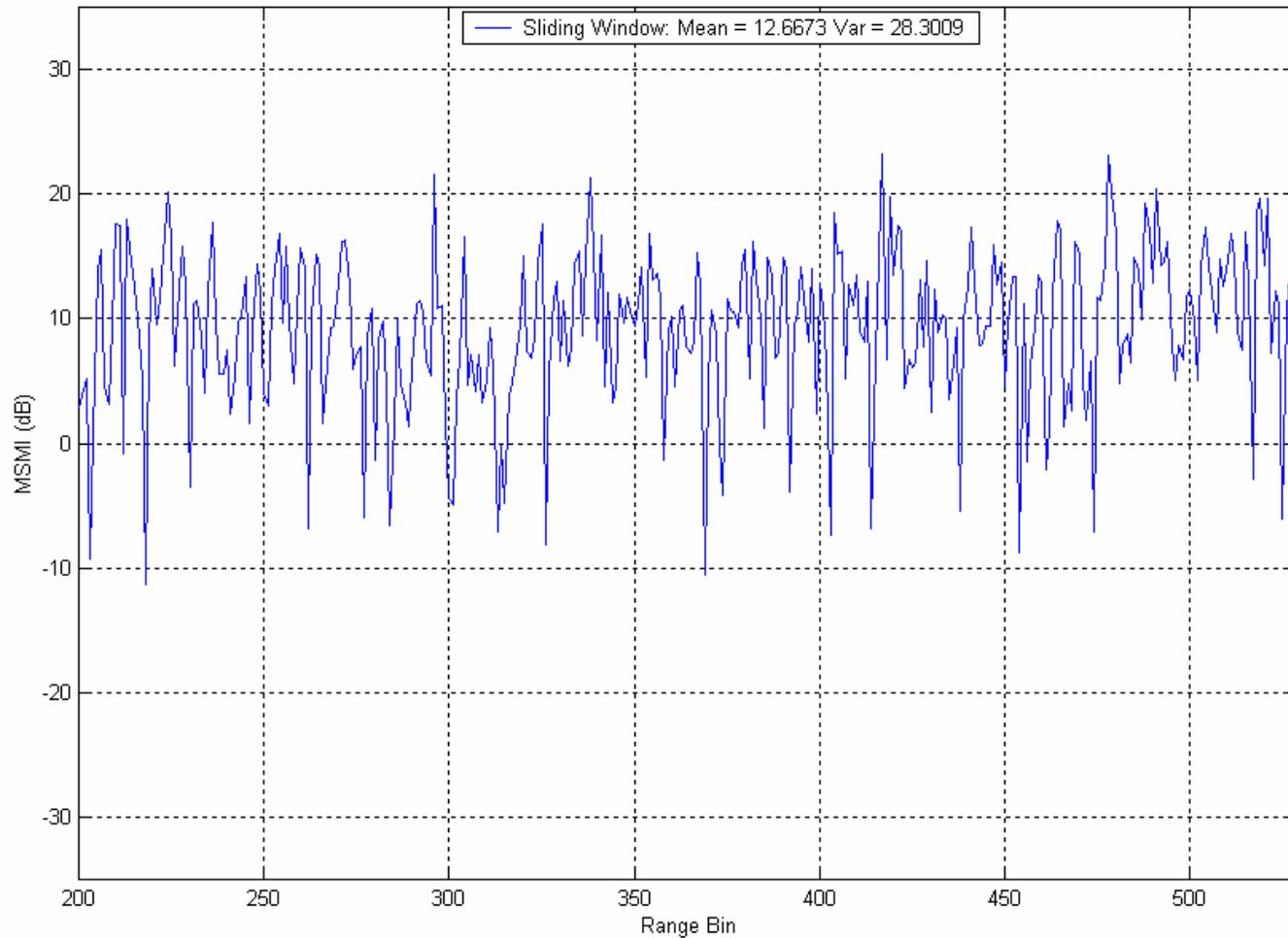


# MSMI Output Using KMapSTAP Algorithm With Injected Target At Range Bin 375



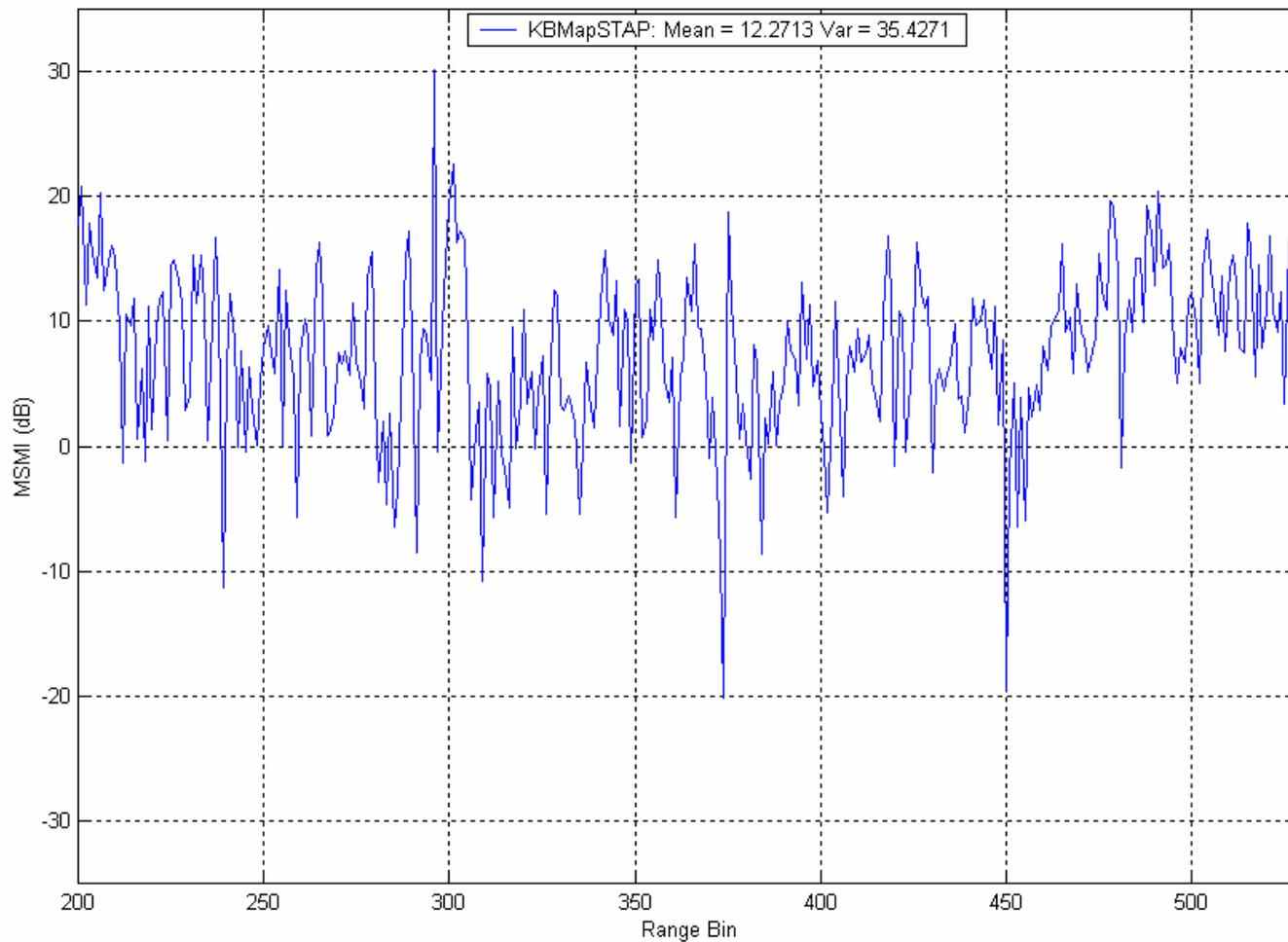


# MSMI Output Using Sliding Window Algorithm With Injected Target At Range Bin 296





# MSMI Output Using KMapSTAP Algorithm With Injected Target At Range Bin 296





# Early Assessment

- Outperformed standard windowing approach within heterogeneous environments (up to 9 dB improvement)
- Performed the same as standard windowing approach within homogeneous environments
- However, there are issues with this approach
  - Map data accuracy – data are not always current
  - Digital elevation data needed in mountainous terrains– shadowing effects
  - Weather data is time dependent
  - Time of year – e.g. snow covered terrain
  - Registration and calibration errors must be assessed
  - Variability in STAP results compared to sliding window
- Need to “see what the radar is seeing”
- Map data is necessary but not sufficient for filtering and detection– also need mapping data for tracking (e.g. roads and railroads)

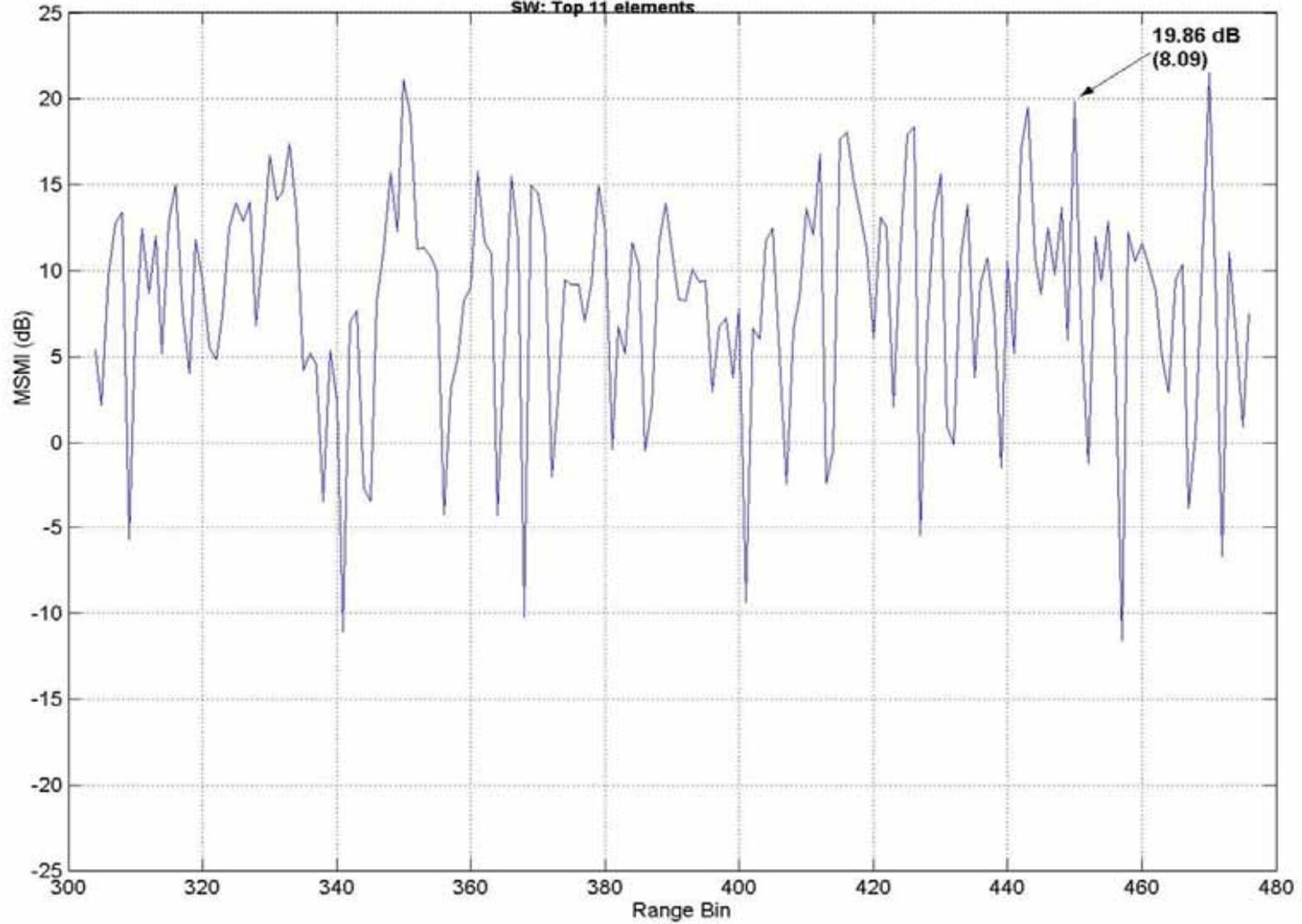


# Secondary Data Guard Cells

- Analysis of early results were suspect
- Analysis of Moving Target Simulated (MTS) data indicated range-Doppler spread
  - Potentially violating i.i.d. criteria in STAP processing
- Excluded range cells around secondary data cells to mitigate the affects of range-Doppler spreading
- Due to number of required range cells, total number of range cells available, and the number of guard cells:
  - Tried to use full array (22 elements) – Not enough samples
  - Used upper row array (11 elements)

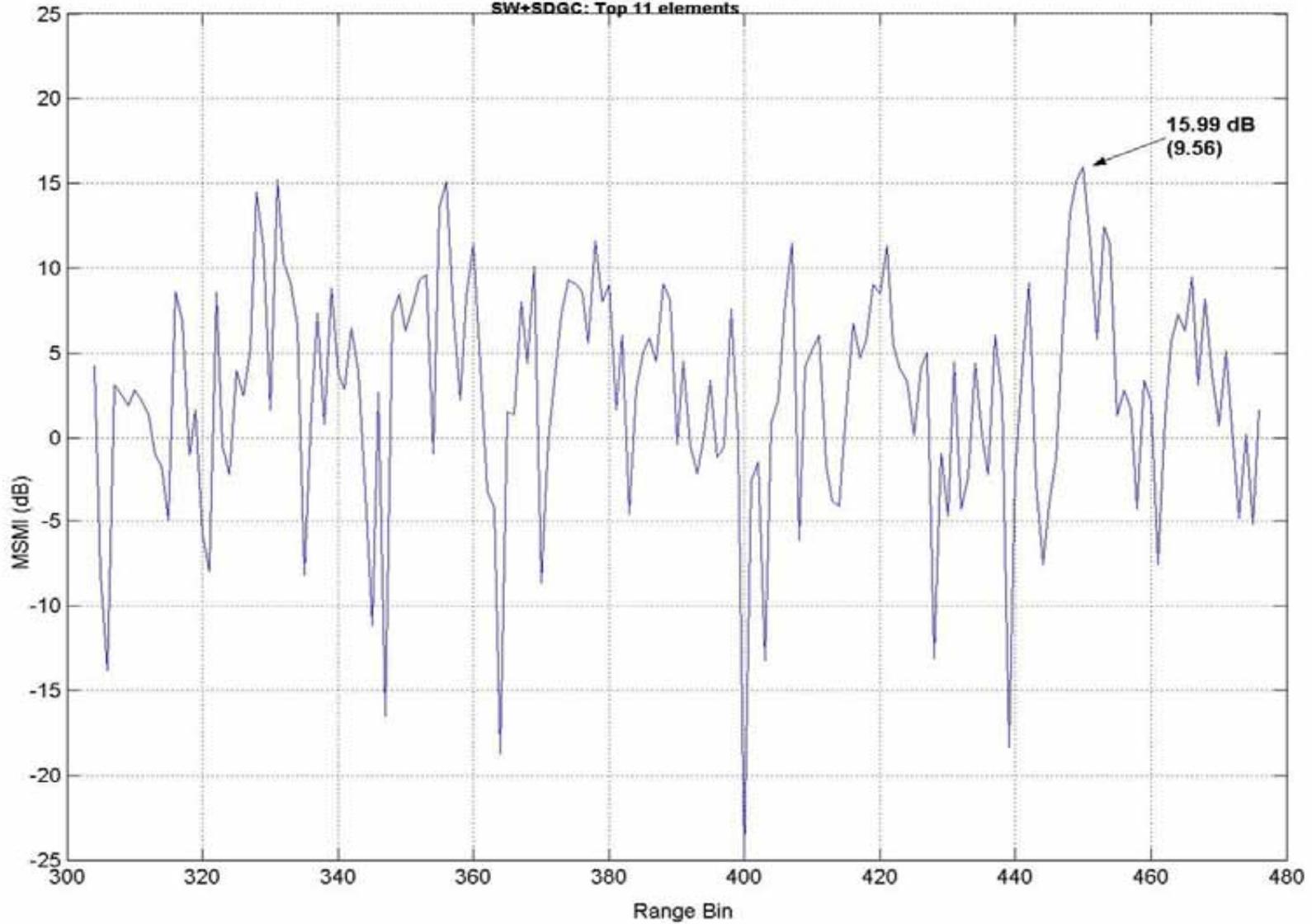


MCARM Flight 5, Acq 151  
MSMI: mean=11.7705 std=13.2673 ang=65 dop=65 gc=2  
0Hz MTS Chirp  
SW: Top 11 elements



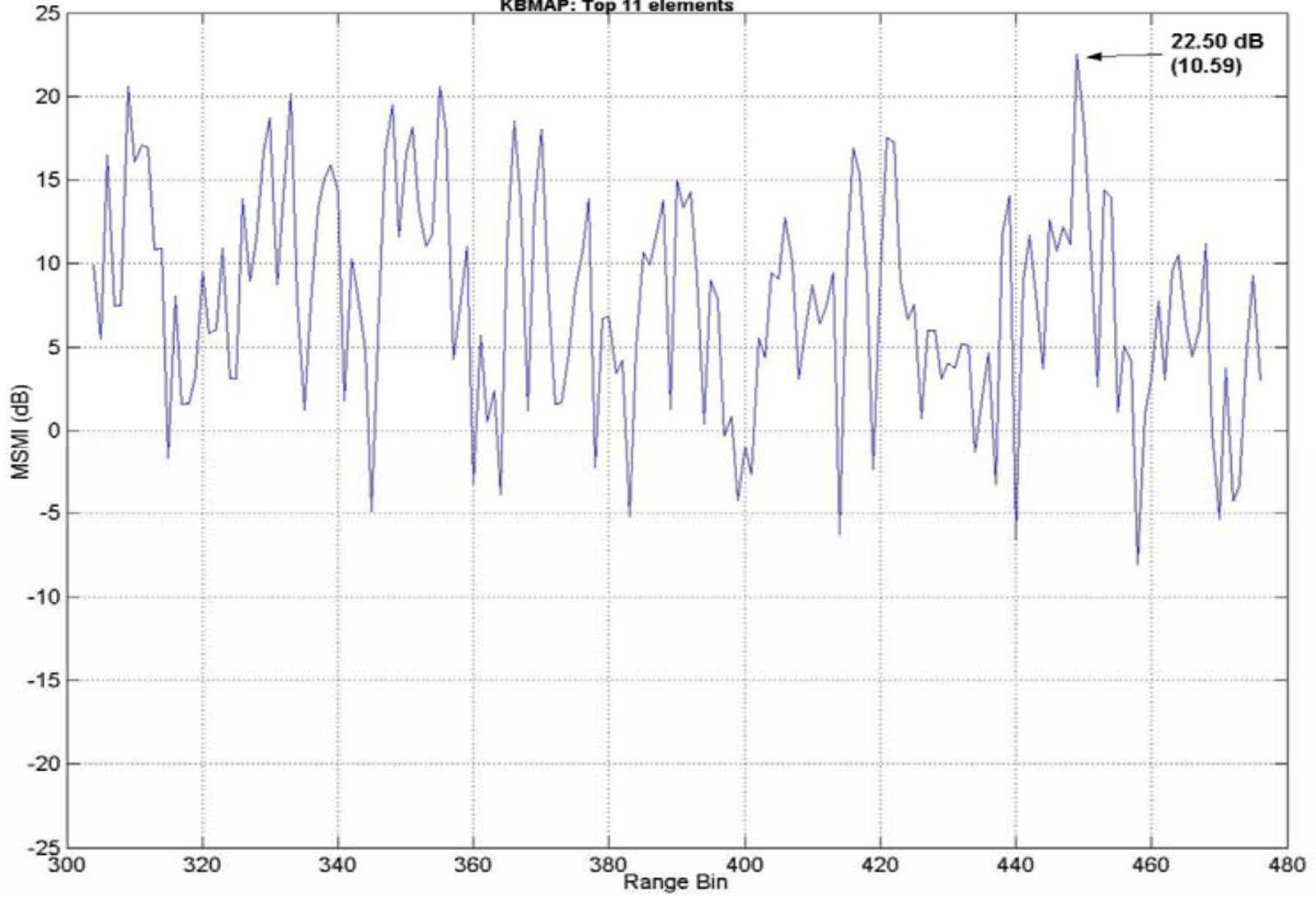


MCARM Flight 5, Acq 151  
MSMI: mean=6.4329 std=8.09 ang=65 dop=65 gc=2  
0Hz MTS Chirp  
SW+SDGC: Top 11 elements



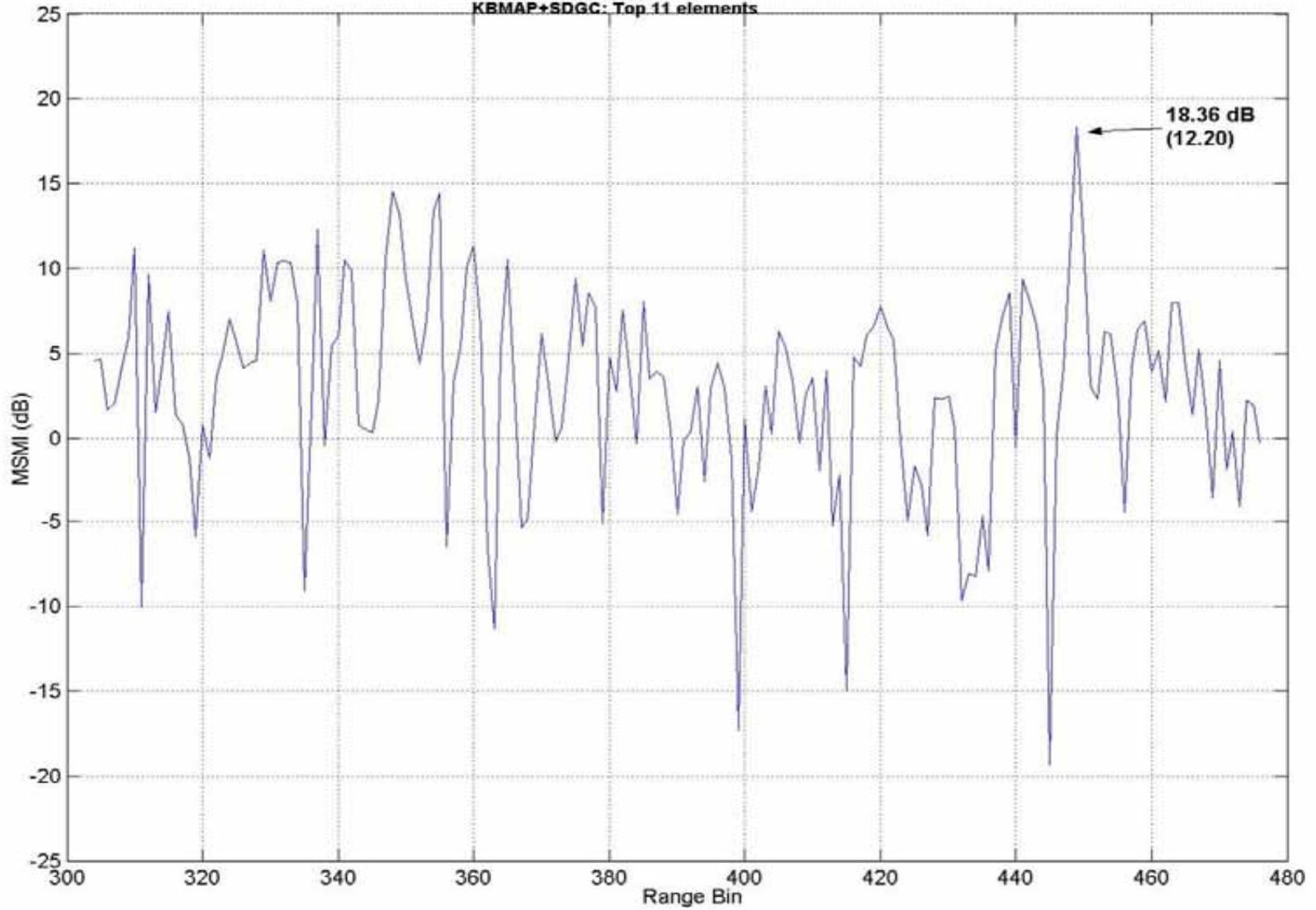


MCARM Flight 5 Acq 151  
MSMI: mean=11.9114 std=13.9276 ang=65 dop=65 gc=2  
0Hz MTS Chirp  
KBMAP: Top 11 elements



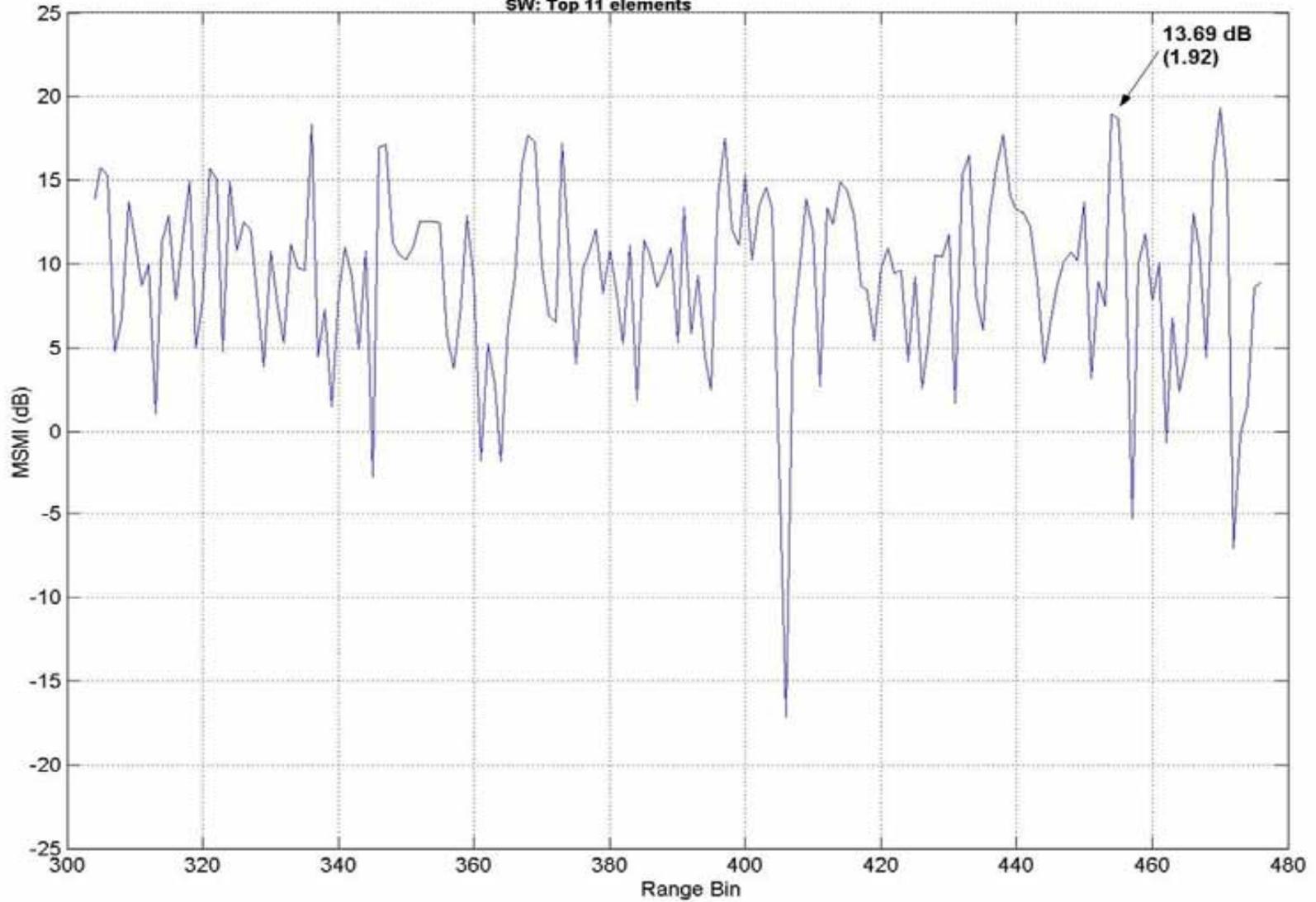


MCARM Flight 5, Acq 151  
MSMI: mean=6.1616 std=8.3259 ang=65 dop=65 gc=2  
0Hz MTS Chirp  
KMAP+SDGC: Top 11 elements



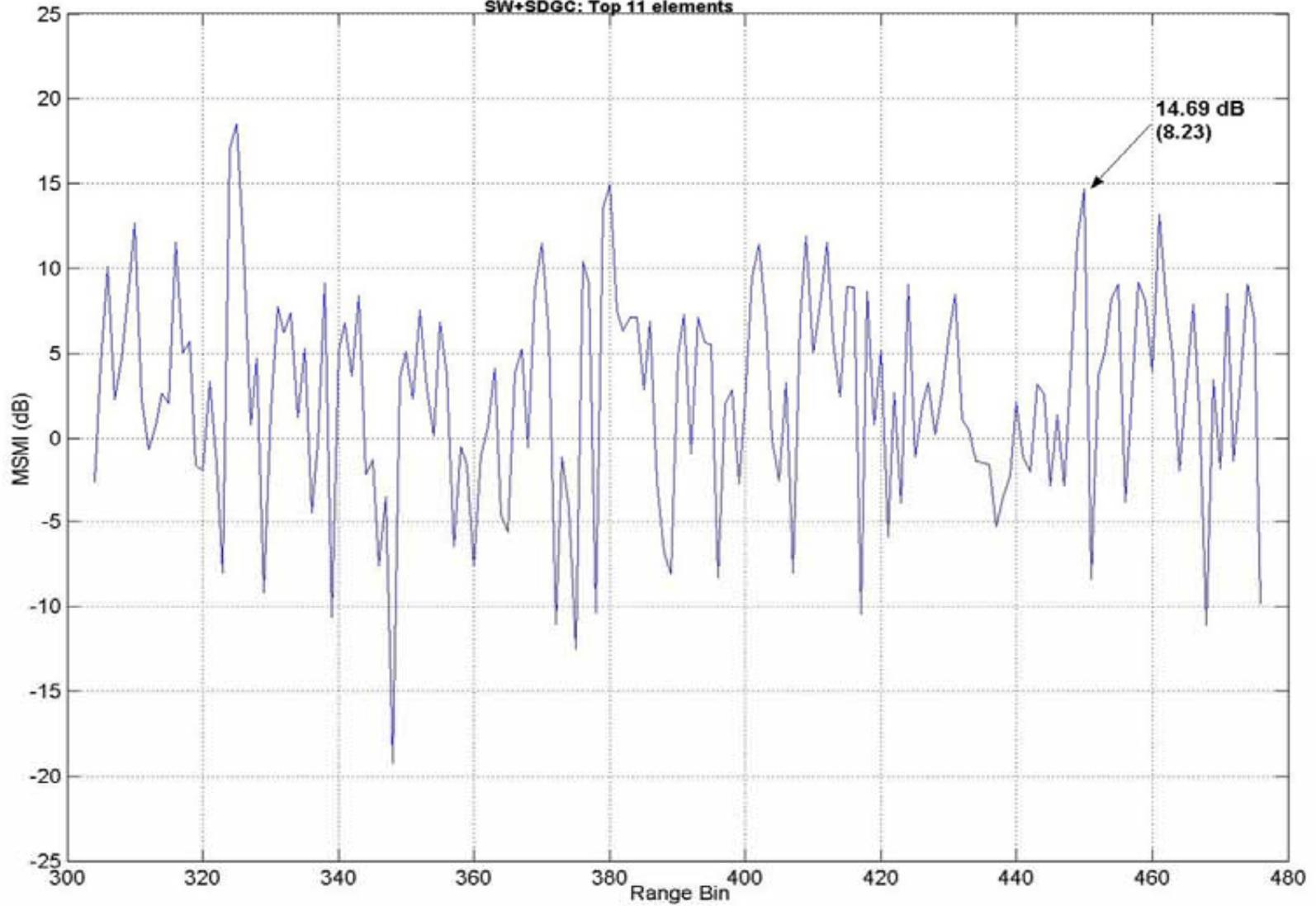


MCARM Flight 5 Acq 152  
MSMI: mean=11.7735 std=12.0242 ang=85 dop=52 gc=2  
-200Hz MTS Chirp  
SW: Top 11 elements



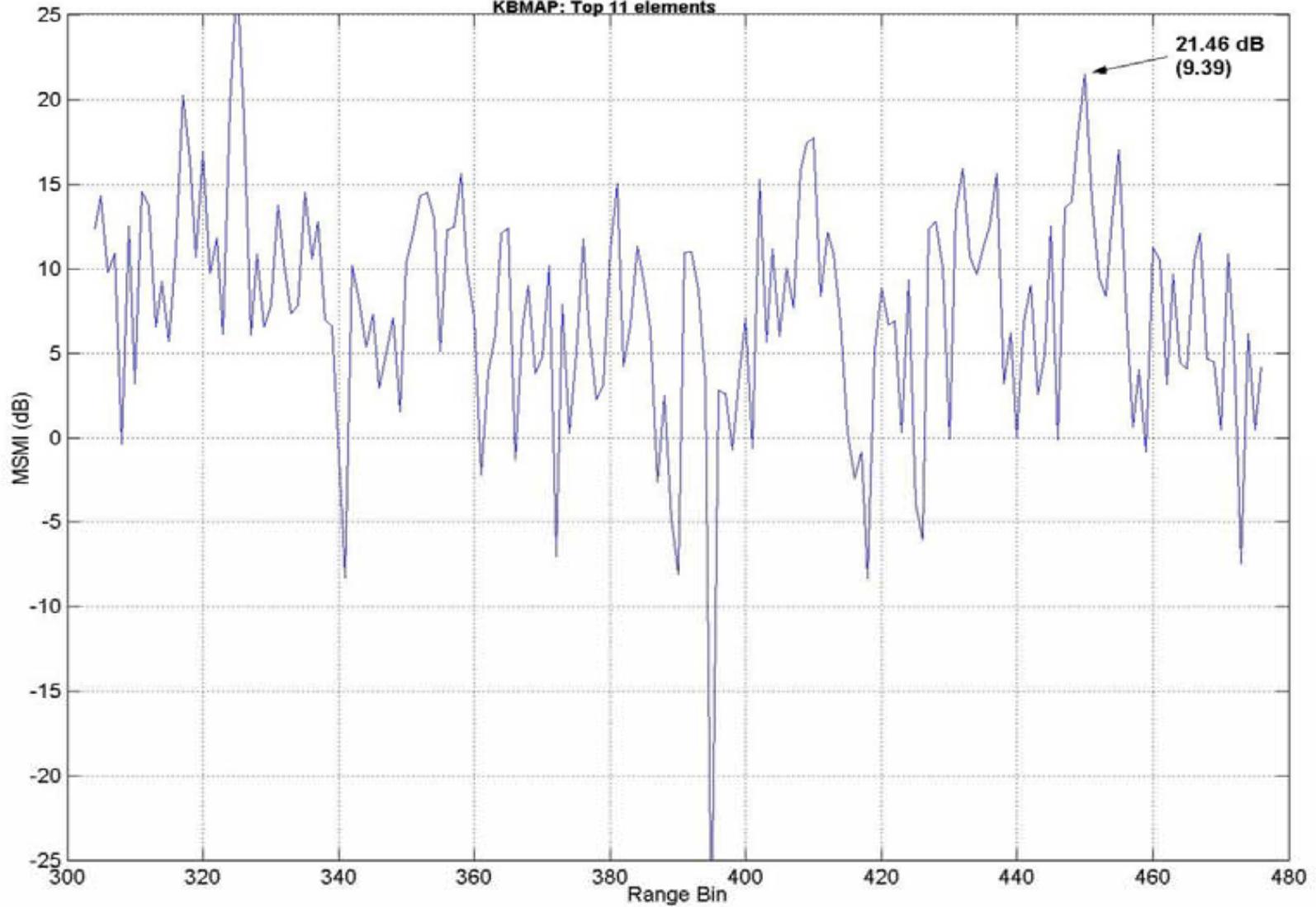


MCARM Flight 5 Acq 152  
MSMI: mean=6.4557 std=9.0238 ang=85 dop=51 gc=2  
-200Hz MTS Chirp  
SW+SDGC: Top 11 elements



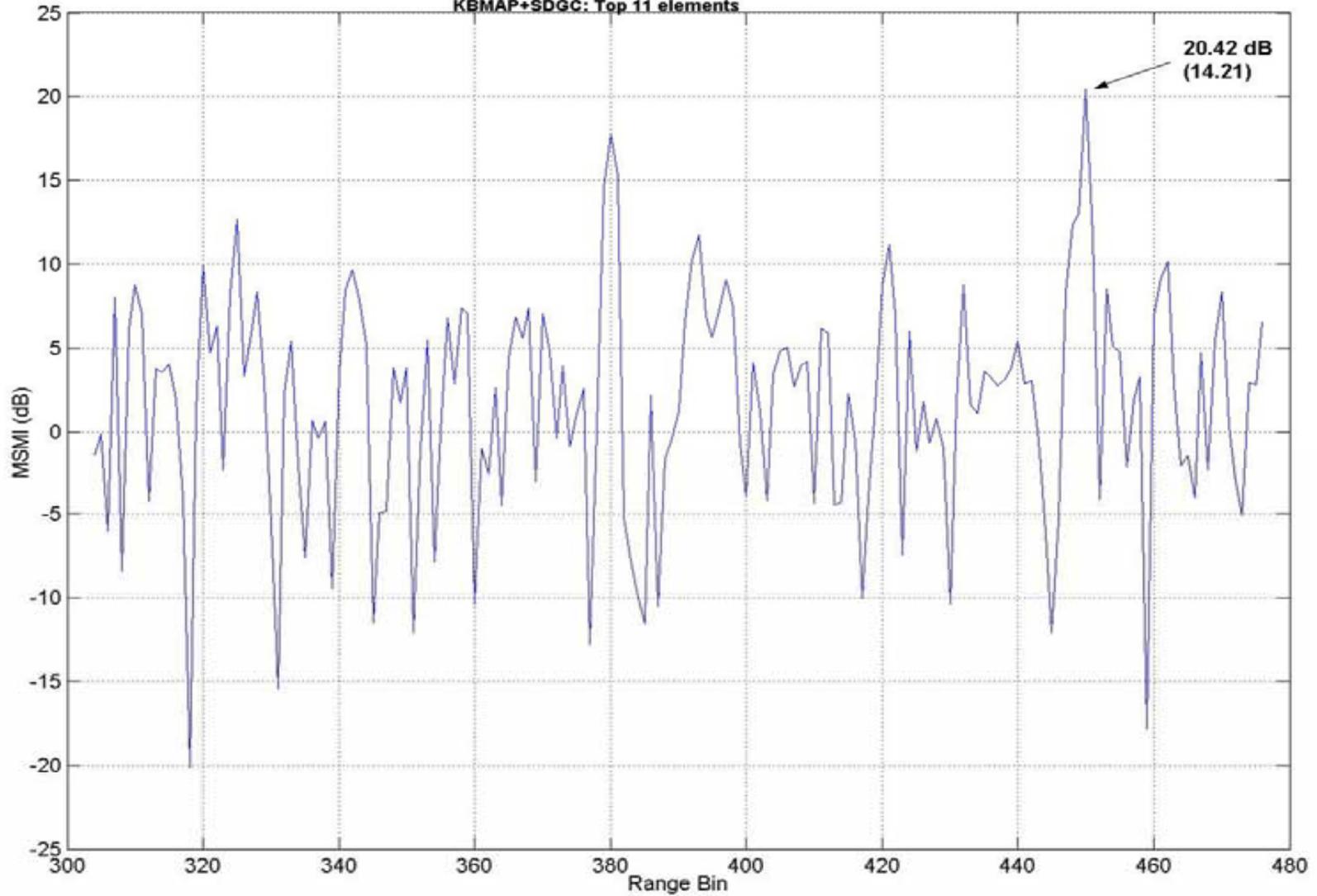


MCARM Flight 5 Acq 152  
MSMI: mean=12.0737 std=16.8982 ang=85 dop=51 gc=2  
-200Hz MTS Chirp  
KBMAP: Top 11 elements



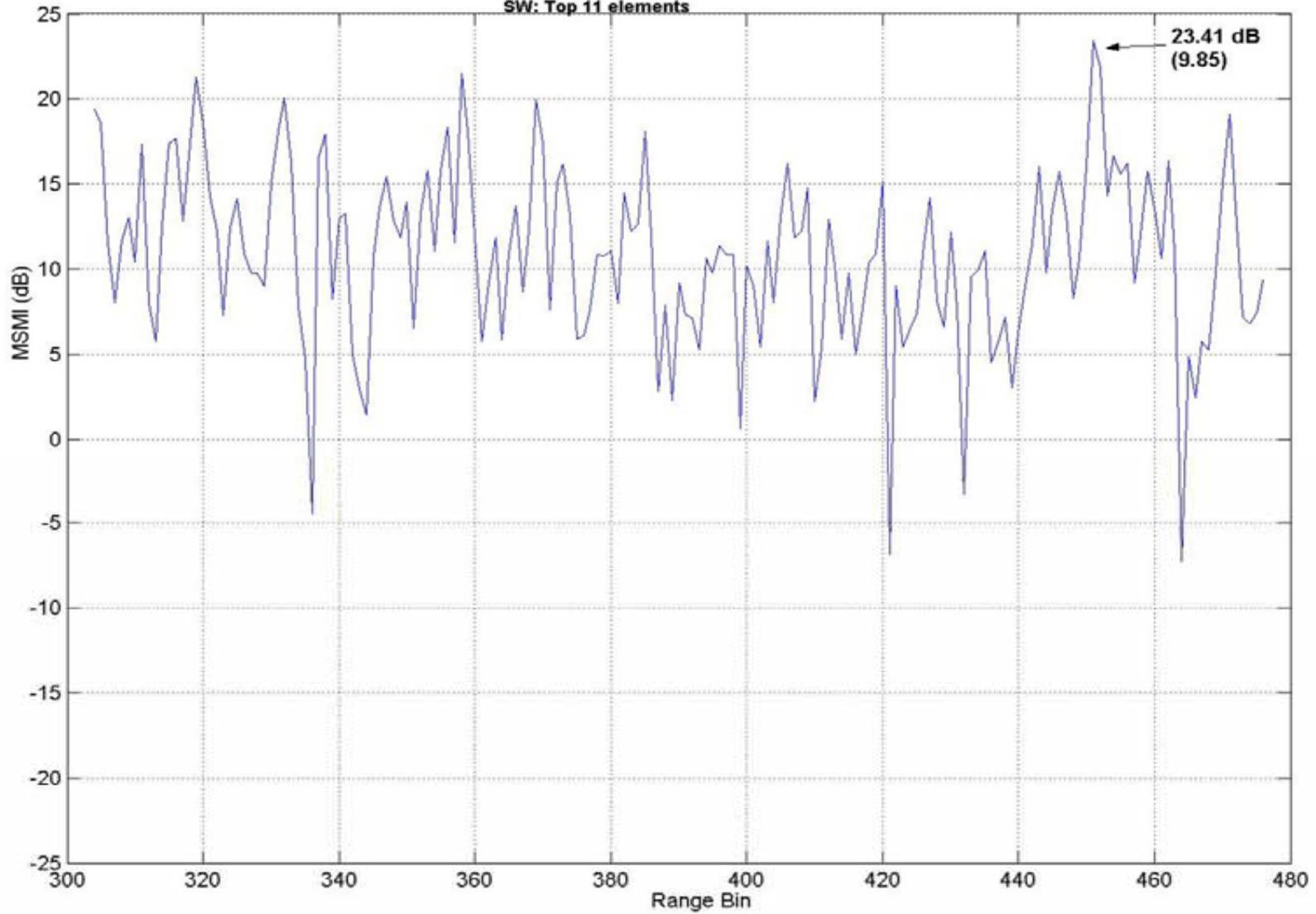


MCARM Flight 5 Acq 152  
MSMI: mean=6.2127 std=10.1274 ang=85 dop=51 gc=2  
-200Hz MTS Chirp  
KBMAP+SDGC: Top 11 elements



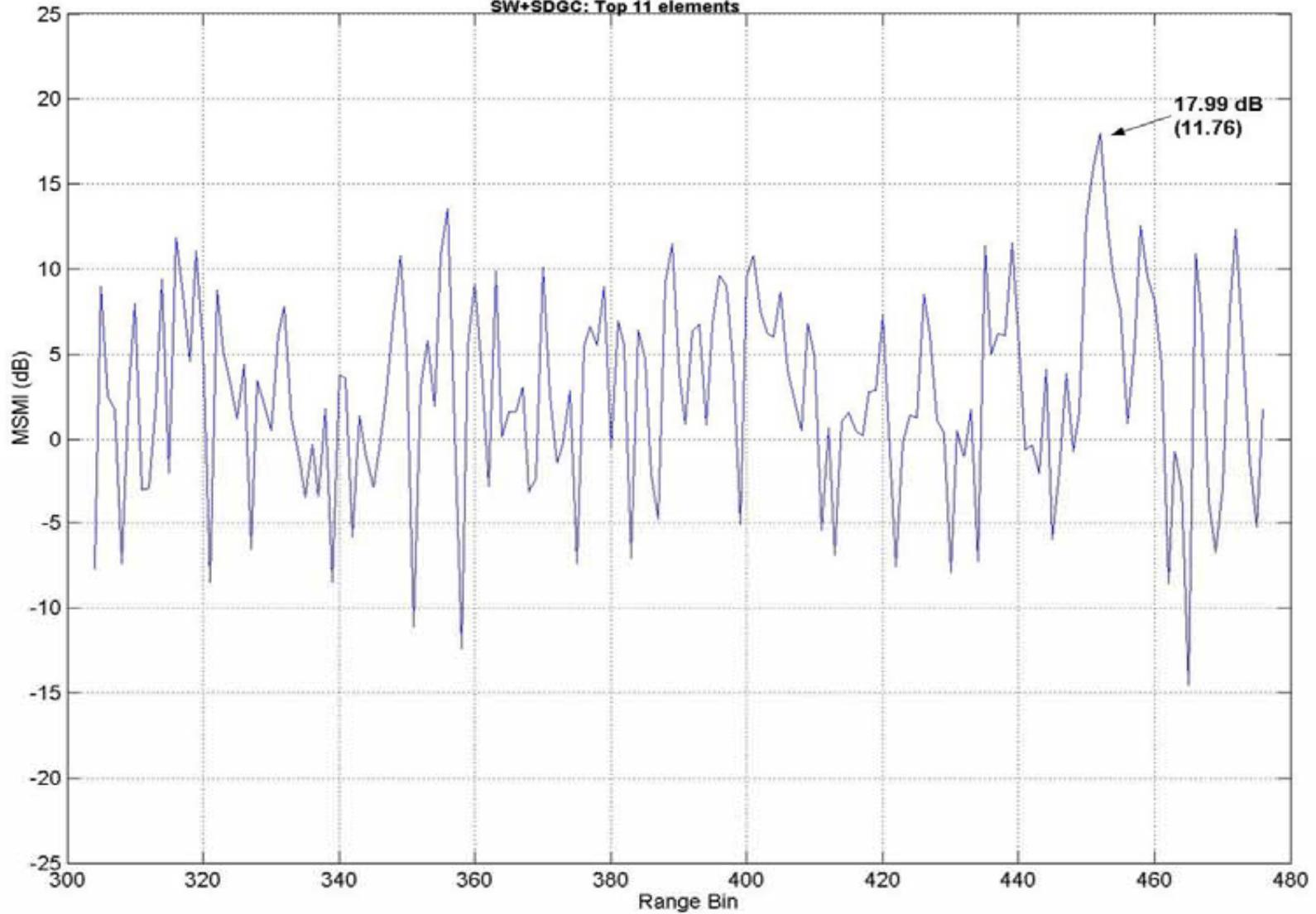


MCARM Flight 5 Acq 153  
MSMI: mean=13.556 std=14.7336 ang=85 dop=52 gc=2  
-200Hz MTS Chirp  
SW: Top 11 elements



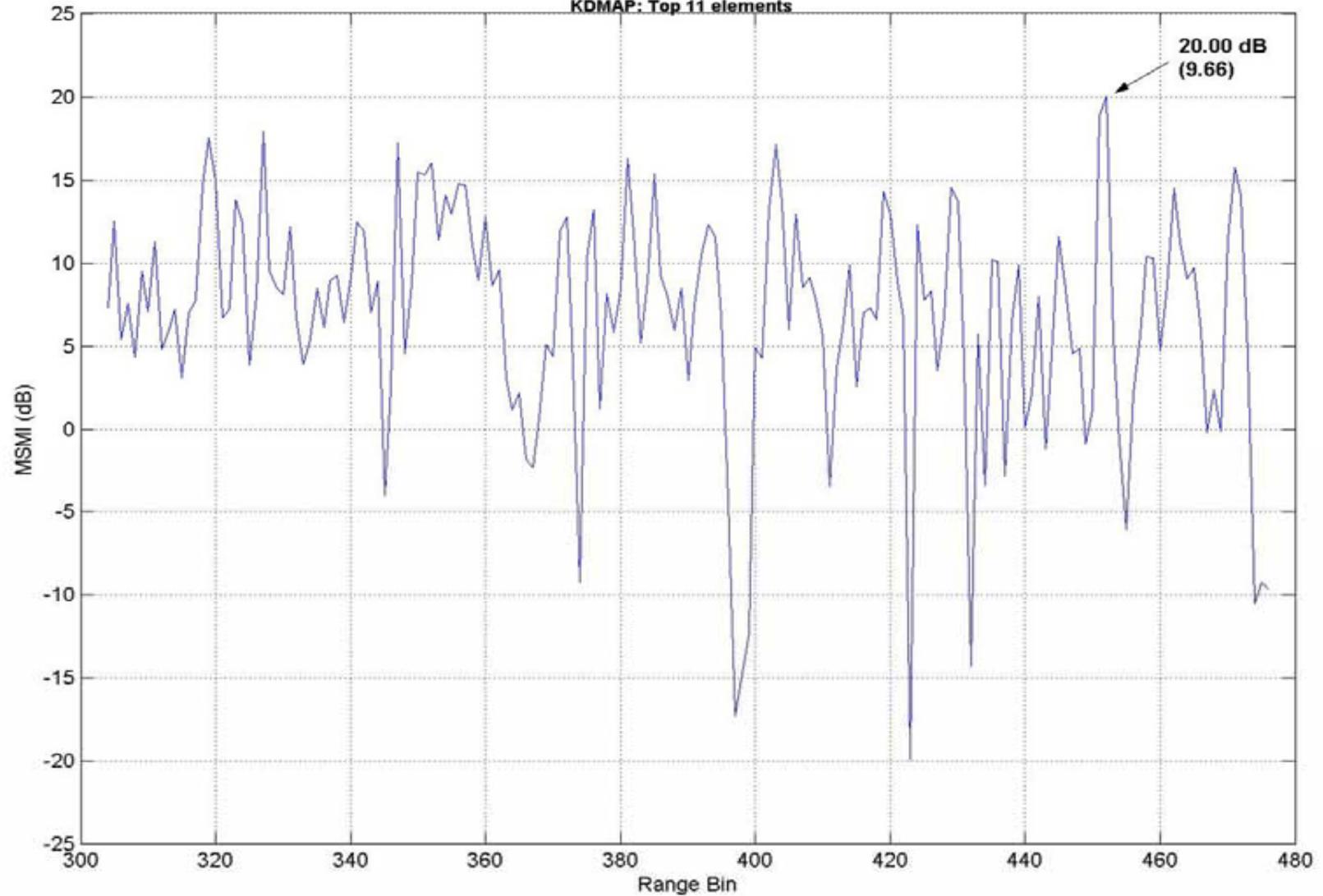


MCARM Flight 5 Acq 153  
MSMI: mean=6.2266 std=8.3426 ang=85 dop=52 gc=2  
-200Hz MTS Chirp  
SW+SDGC: Top 11 elements



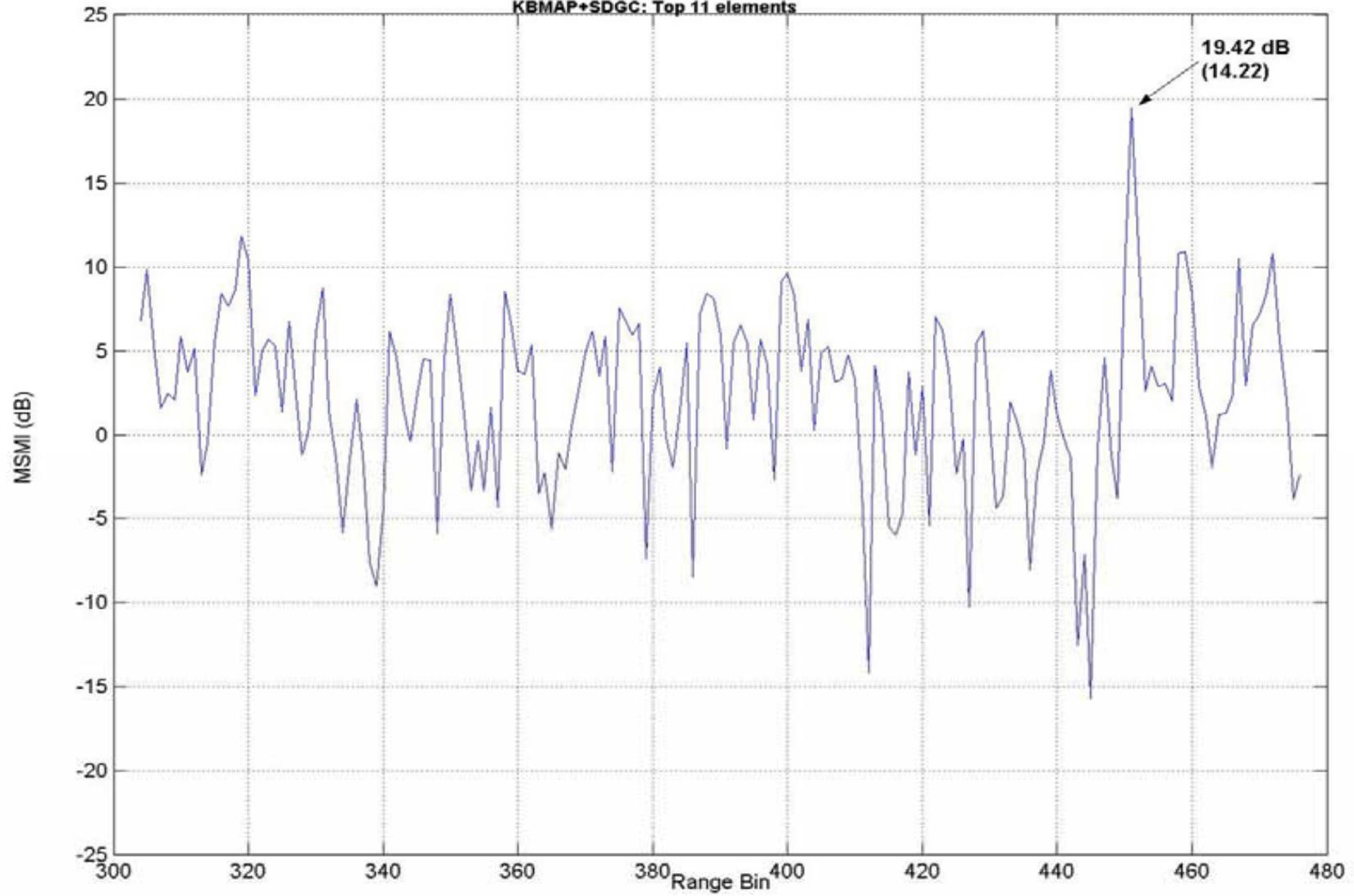


MCARM Flight 5 Acq 153  
MSMI: mean=10.3445 std=11.5531 ang=85 dop=52 gc=2  
-200Hz MTS Chirp  
KDMAP: Top 11 elements





MCARM Flight 5 Acq 153  
MSMI: mean=5.2005 std=8.4797 ang=85 dop=52 gc=2  
-200Hz MTS Chirp  
KMAP+SDGC: Top 11 elements





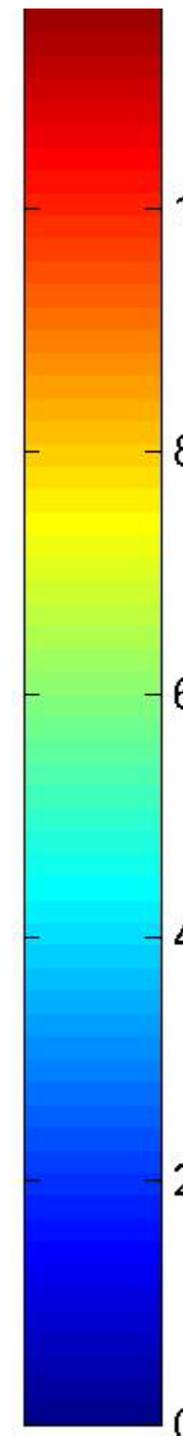
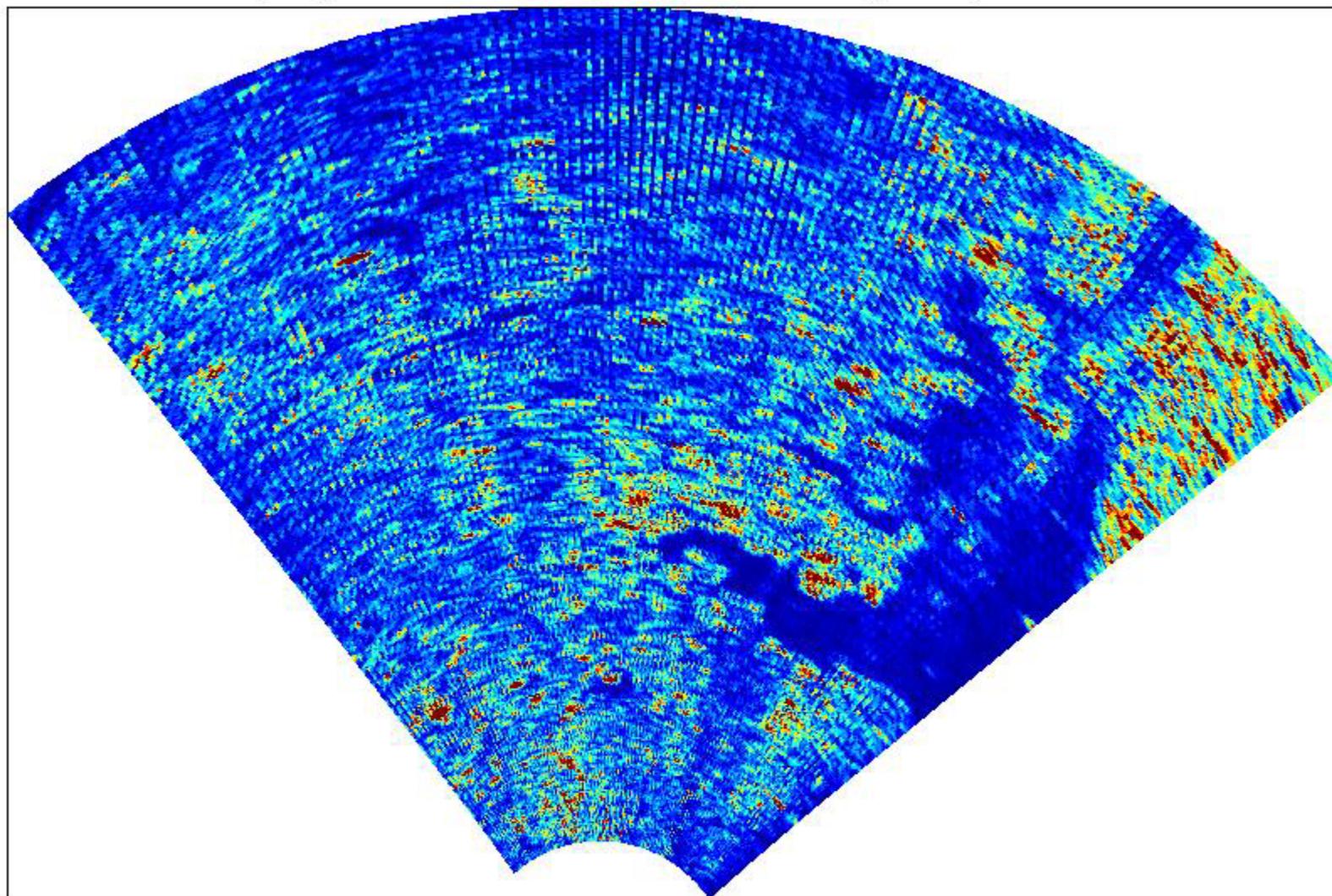
# Results

- Knowledge-aided approach performed 3-7 dB better than cell averaging symmetric (sliding window) method in non-homogeneous terrain environments
- Used modified sample matrix inversion (MSMI) – as our test statistic
- MTS target at range bin 450
- Ratio of MSMI of target to average MSMI, over all ranges, is our preferred performance measure (PPM)



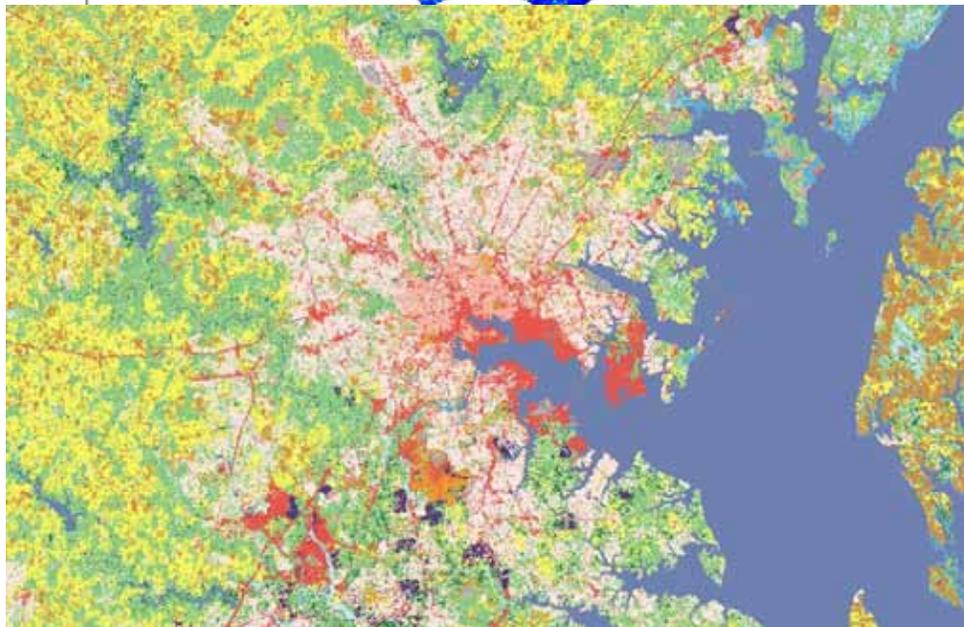
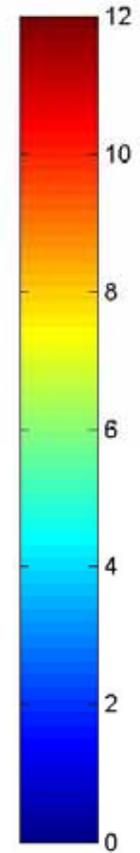
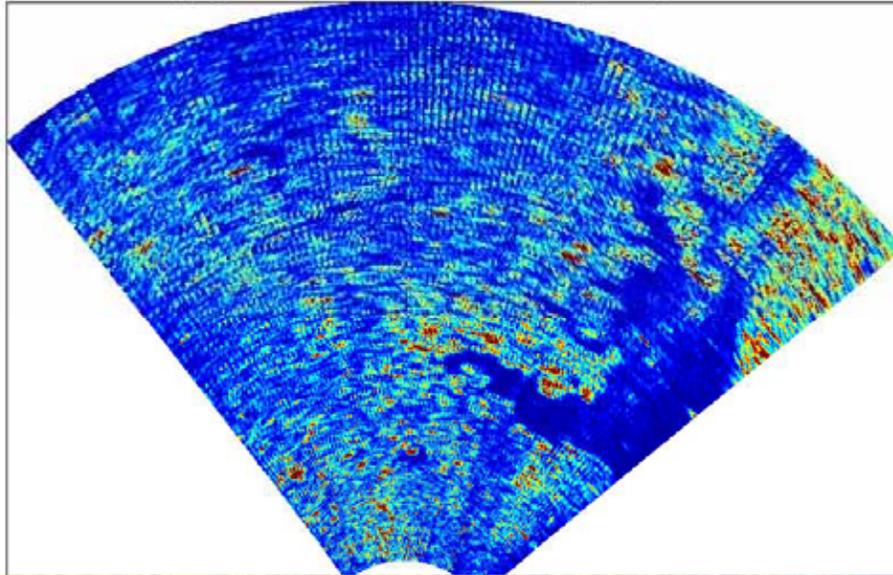
# Seeing What the Sensor Sees

**Composite Map Generated from Three MCARM Radar CPIs  
Flight 5, Acq. 150, 153, and 157  
(Projected in Albers Equal-Area Conic Projection)**





Composite Map Generated from Three MCARM Radar CPIs  
Flight 5, Acq. 150, 153, and 157  
(Projected in Albers Equal-Area Conic Projection)





# Digital Elevation Model (DEM) Data



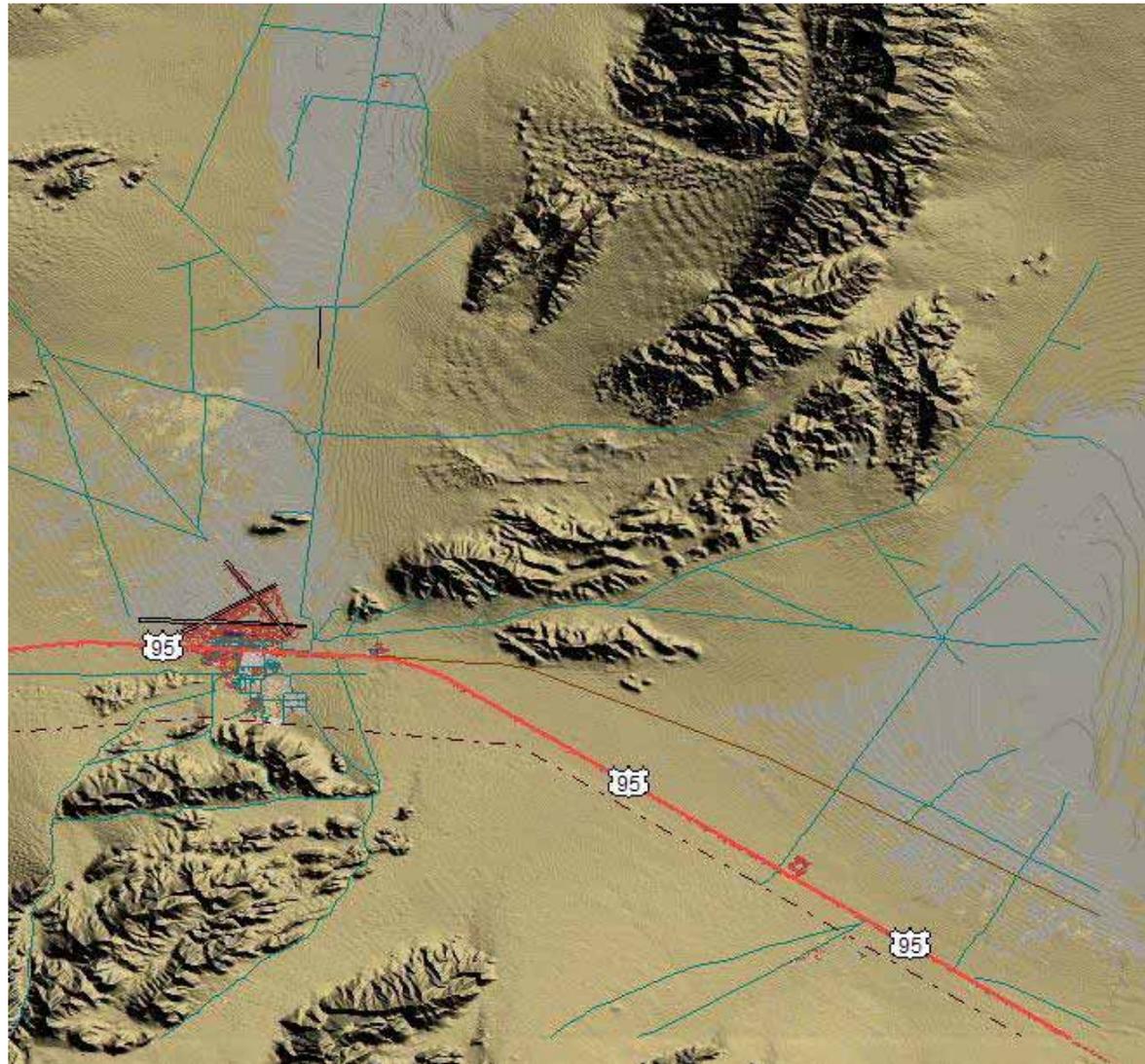
# DEM Data Algorithm Development



- Chose a mountainous region
- A real radar with multiple CPIs
- Real truth data – more than MTS data
- Minimal variation in NLCD data – not included in algorithm
- Algorithm is automated in MATLAB (based upon % shadowing and reflection angle statistics per range ring)

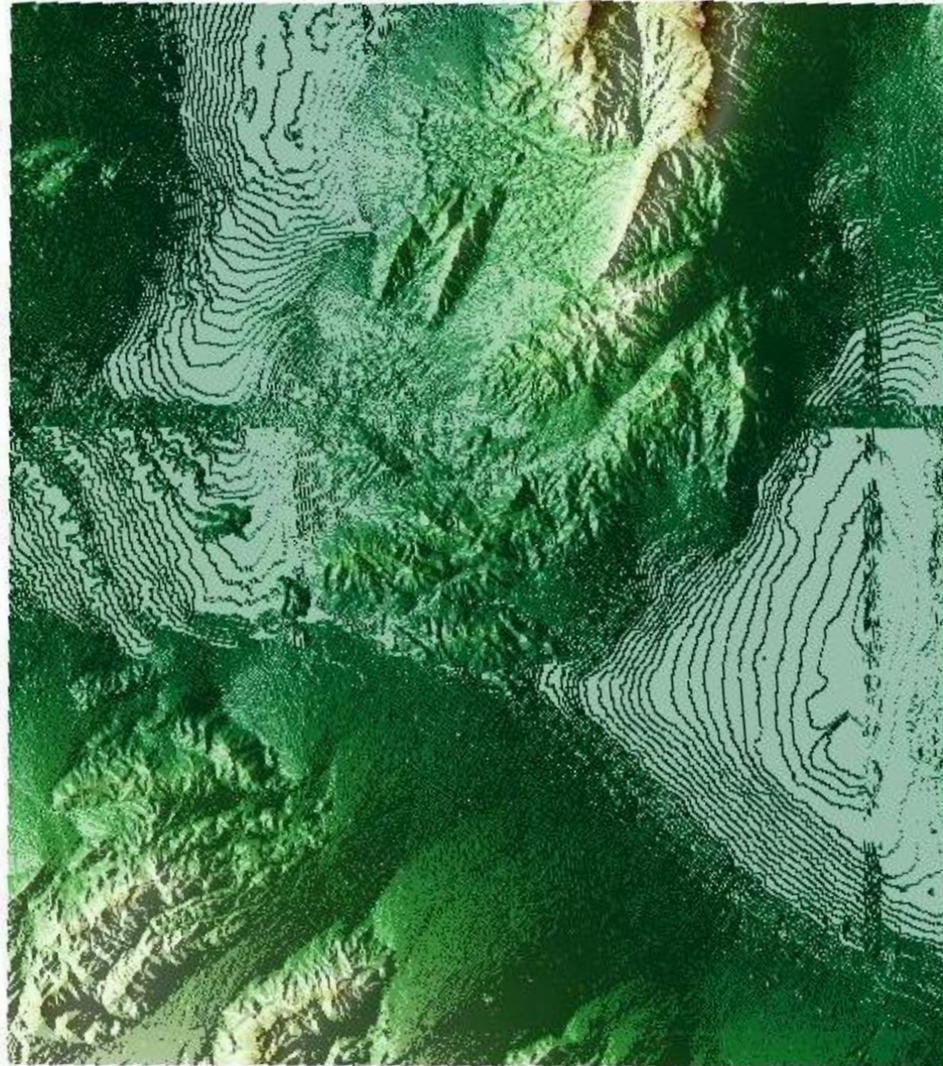


## Scene with DEM, LULC and DLG Terrain Data



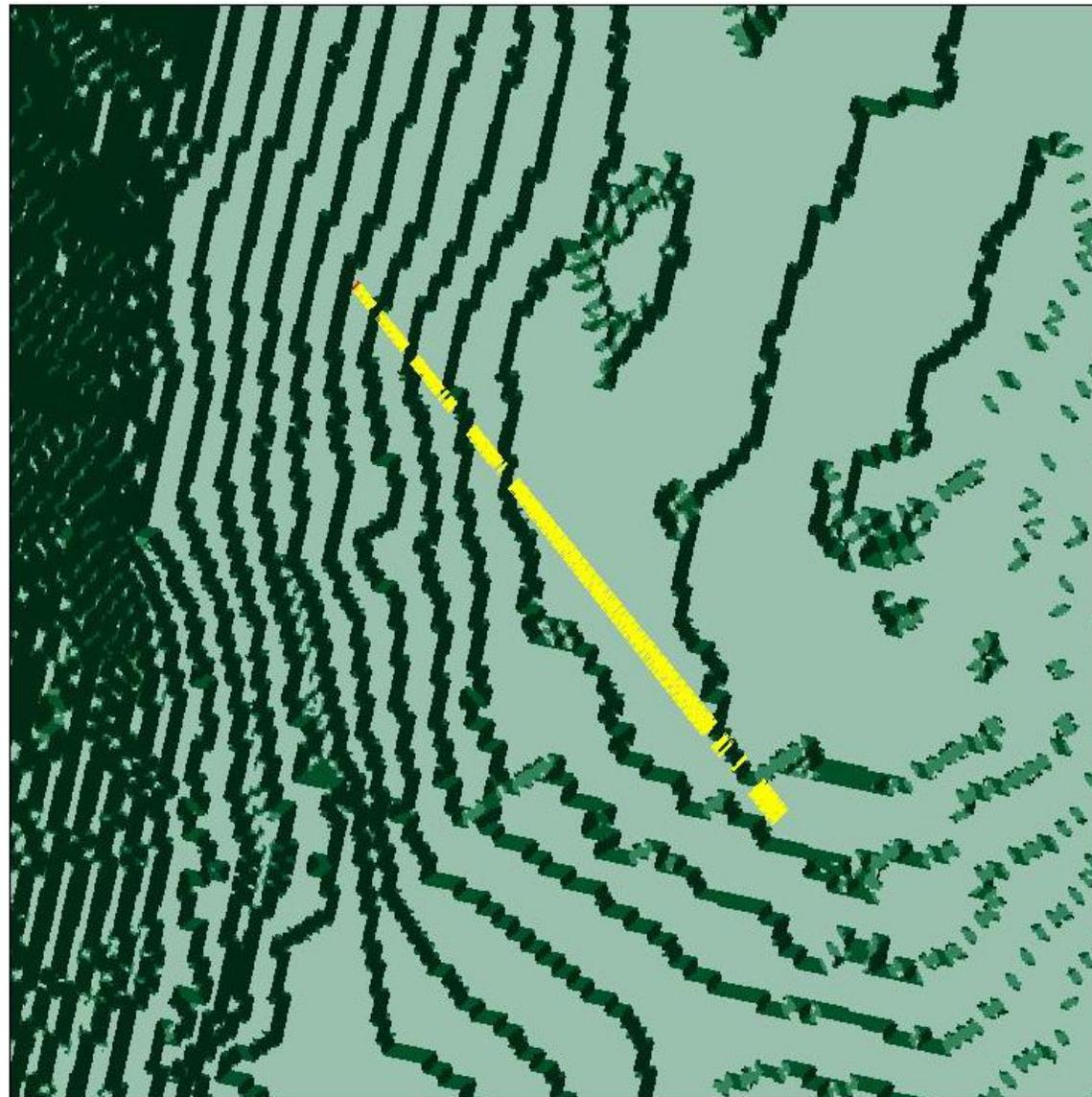


## Scene with 10 Meter DEM Terrain Data in MATLAB



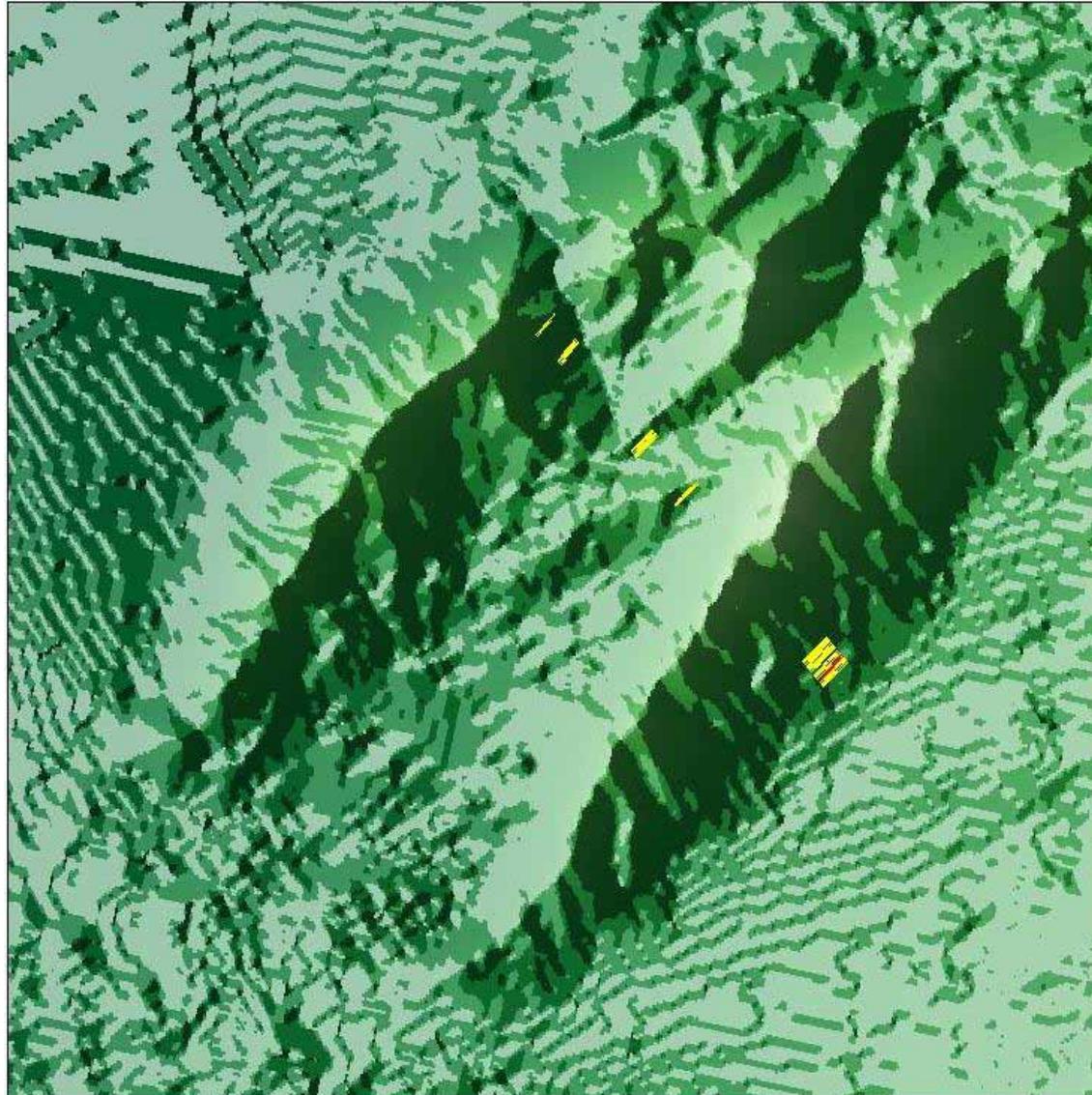


## Chosen Range Bins on Flat Terrain (Test Cell Red, Sample Cells Yellow)



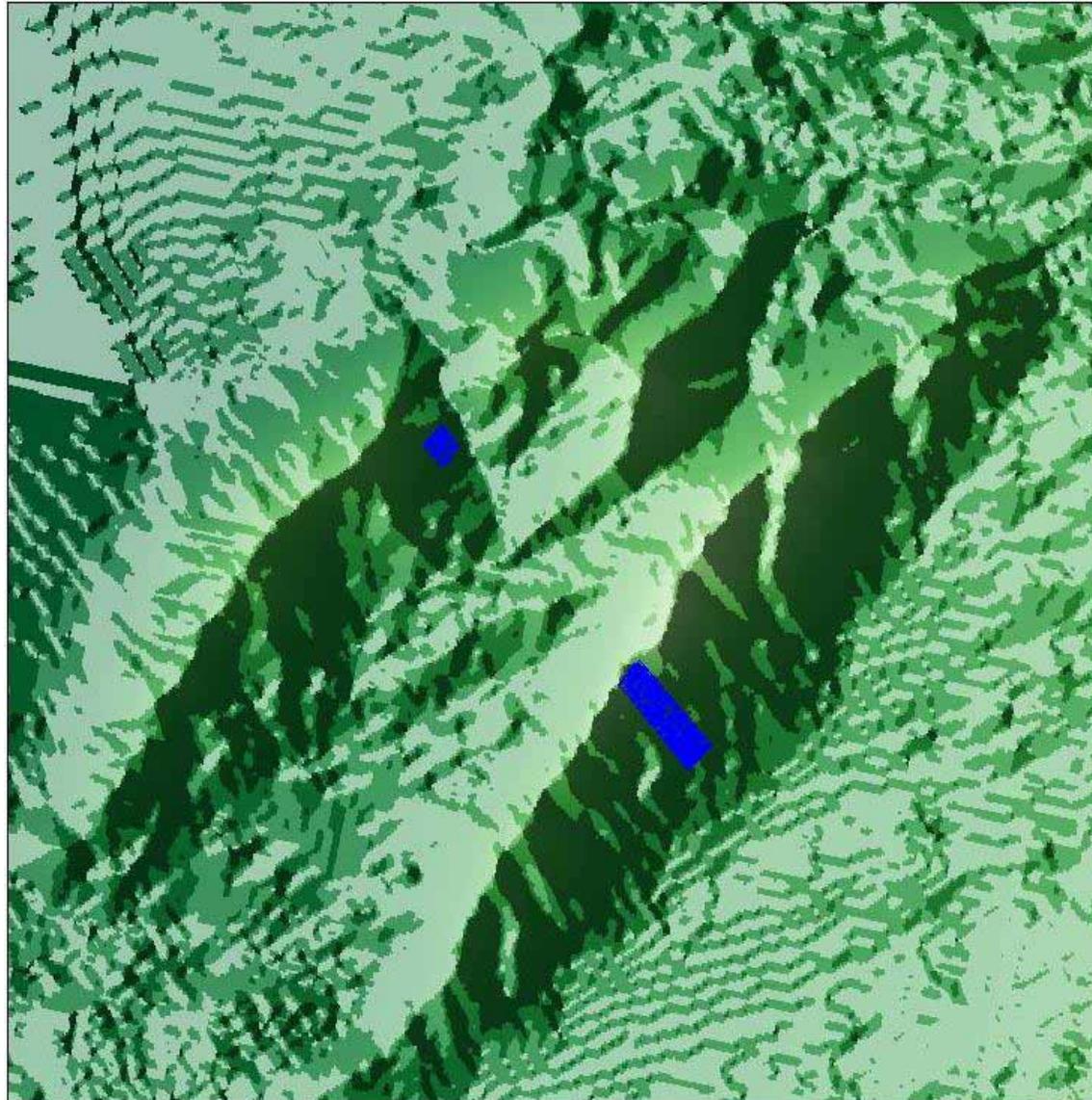


## Chosen Range Bins on Mountainous Terrain (Test Cell Red, Sample Cells Yellow)





## Shadowed Range Cells Shown in Blue

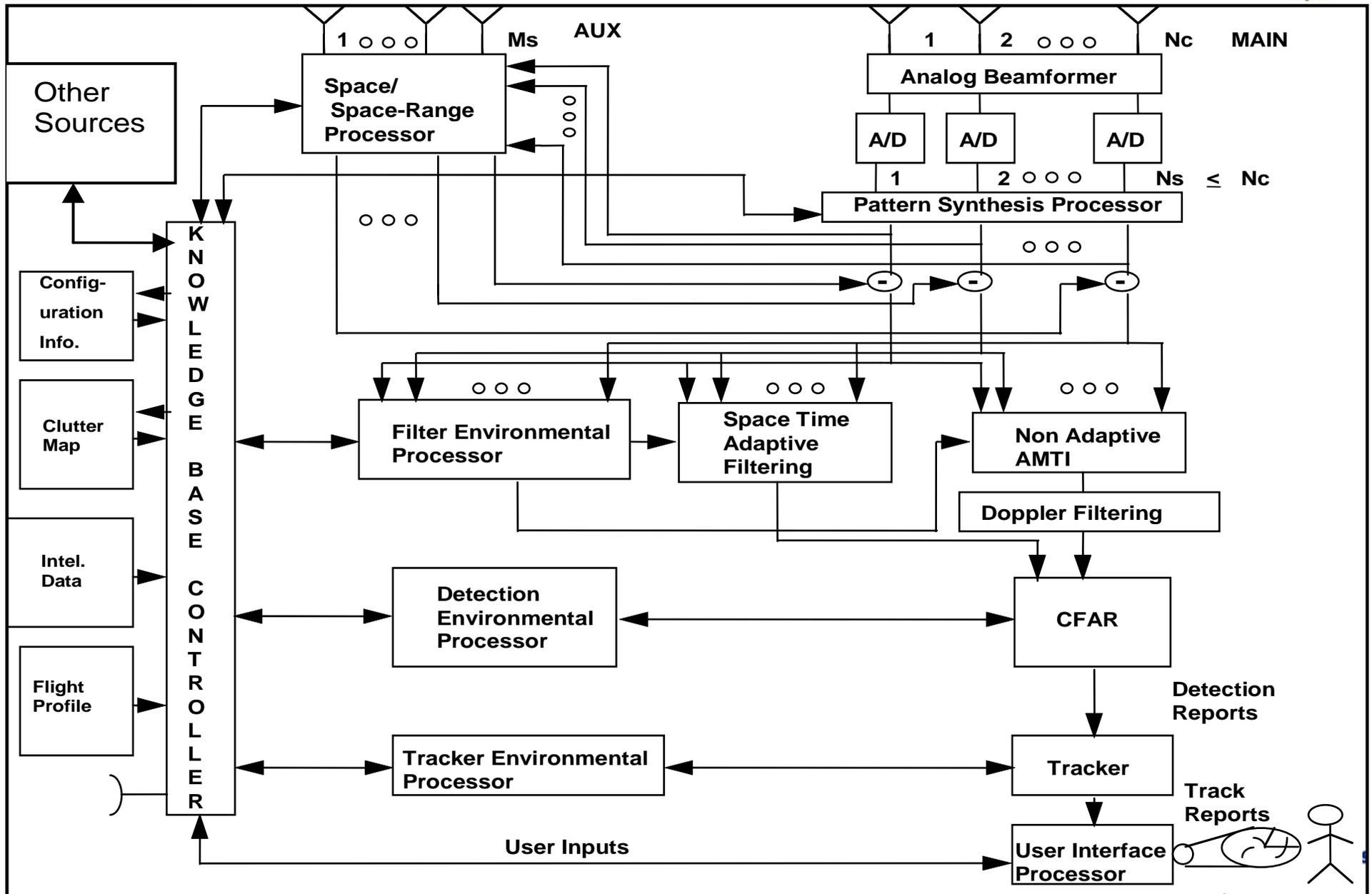




# Agile Intelligent Radar System (AIRS)



# AIRS Architecture



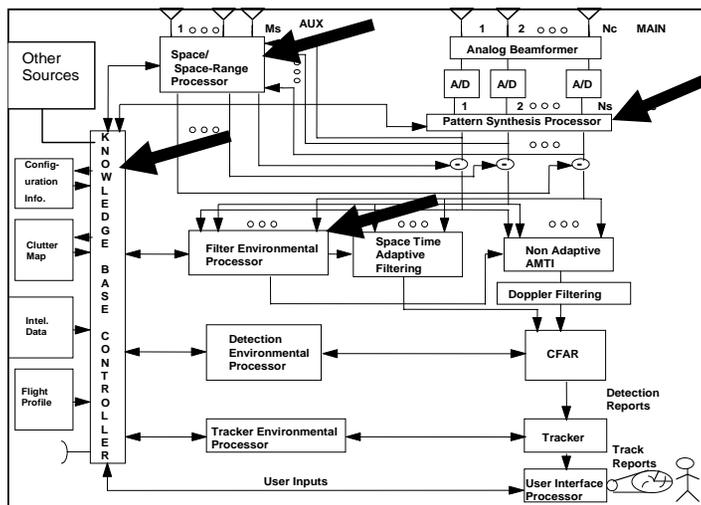


# An Agile Intelligent Radar System (AIRS)

- The KB Signal and Data Processing portion of the Intelligent Sensor System
- KB Controller (KBC), processors, outside data sources, communications, user interface, and pre-loaded data
- Processors work independently and cooperatively
- The KBC handles all interrupts, assigns tasks, manages processors, communicates results, and interfaces with the user



# Space/Space-Range Processor (SSRP), Pattern Synthesis Processor (PSP), Filter Environmental Processor (FEP) and KBC Interfaces

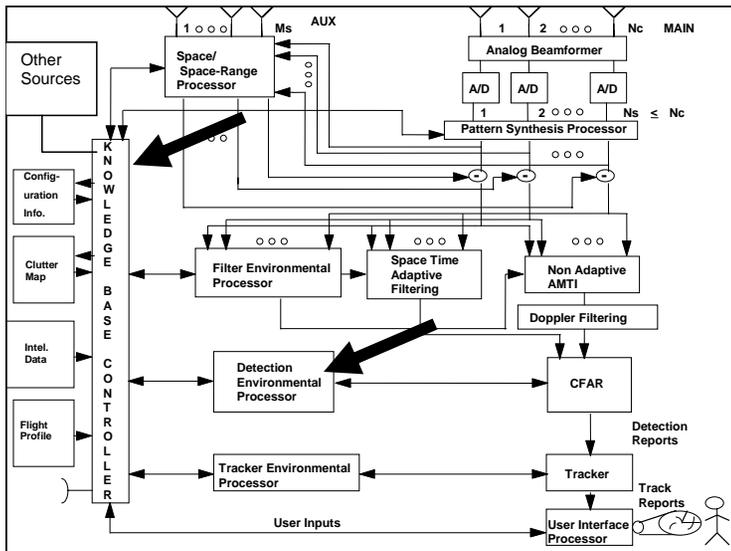


- KBC provides tasking and the location of jammers, discretives, unintentional interferers, etc.
- Processors will “optimize” the KBC tasking
- Data exchange and feedback on results are passed between the processors and the KBC (dB attenuation, gain/loss, algorithms & parameter values)
- KBC will provide control and satisfy operational requests e.g. user wishes to exercise multiple algorithms



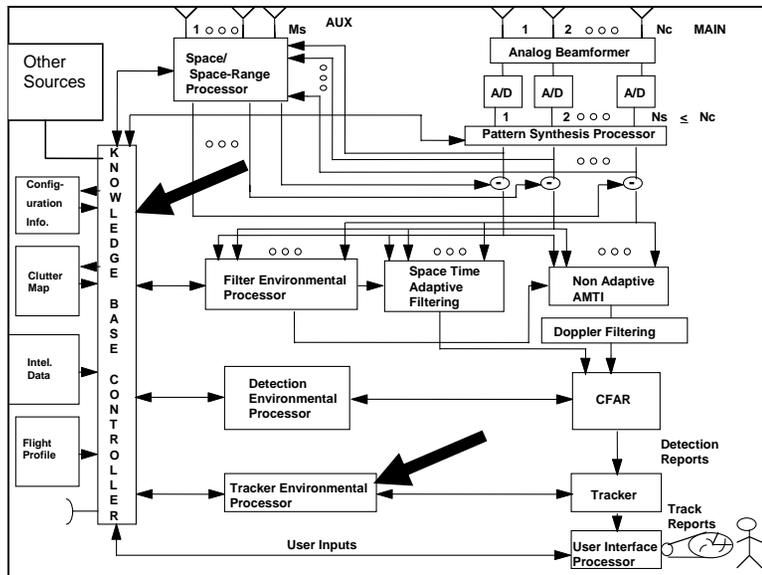
# Detection Environmental Processor (DEP) and KBC Interface

- KBC provides filter data, clutter map and tracker data to the DEP
- DEP uses these data for selecting its CFAR algorithm and setting of thresholds based on “targets” and clutter data
- DEP provides PD, PFA, algorithms, and parameters used
- KBC tasks DEP processing





# Tracker Environmental Processor (TEP) and KBC Interface

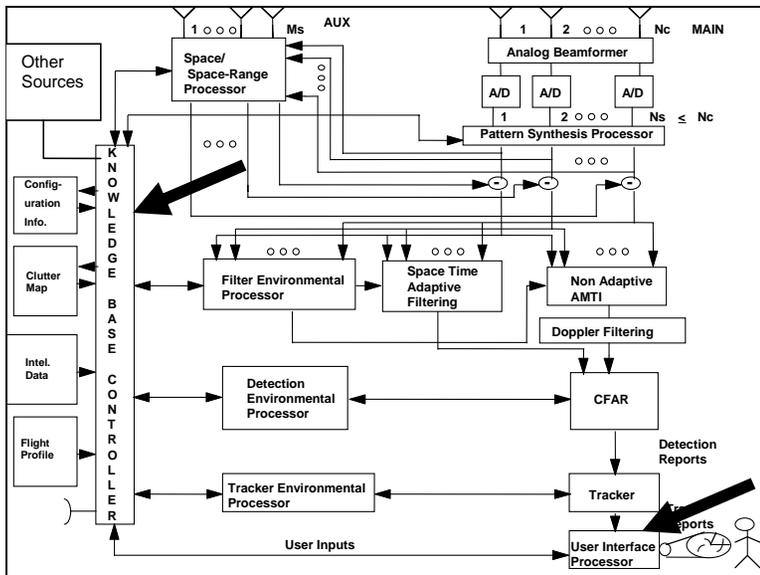


- KBC provides the TEP priority of targets, results of DEP, control and tasking
- TEP uses these data along with terrain data for declaring and managing tracks
- TEP provides all track data (track i.d., track probability or certainty, lost tracks, etc.) algorithms, and parameters used
- KBC tasks TEP processing



# User Interface Processor (UIP) and KBC Interface

- KBC provides the UIP with processed data – intermediate results, performance measures, what and why decisions were made, and assist user in configuring the antenna and processors



- UIP tasks the KBC
- KBC tasks the Process Manager and Data Manager\* to configure all computers & algorithms for next iteration

\*Process Manager and Data Manager are not shown in architecture





# AIRS State Processing

- State 1 – pre-flight loading of intelligence, mission, and terrain data
- State 2 – initial transient state e.g. 0 - 4 CPIs
- State 3 – correlation, performance, assessment and learning state e.g. 4 – 20+ CPIs
- State 4 - steady state e.g. after 1 to 2 flights over area (e.g. race track routes)
- Partitioned KBC into a KB performance processor and a KB control processor



# System States Versus Processors



System States Processors	Pre-Flight	Initiate System & Initial Transient States (4 to 20+ CPIs)	Correlation, Assessment, Learning (1 to 2 Complete Race Tracks of a Defined Scene/Area)	Steady State
<b>K B Performance Processor</b>	1- Locate and load all Potential Discretes, Clutter Boundaries, Shadow Regions, Jammers, Obstacles - Set System Parameters	6- Monitor System	11- Correlate Discretes, Clutter Boundaries, Shadow Regions, Potential Jammers, Obstacles - Evolve Rules - Insert Synthetic Targets - Measure Performances	16- Insert Synthetic Targets - Measure Performances - Change Rule Sets Accordingly
<b>K B Controller</b>	2- Locate and load all Potential Discretes, Clutter Boundaries, Shadow Regions, Jammers, Obstacles - Set System Parameters	7- Initiate System and Monitor	12- Correlate Discretes, Clutter Boundaries, Shadow Regions, Potential Jammers, Obstacles - Evolve Rules	17- Measure Performances - Change Rule Set Accordingly



# System States Versus Processors (Continued)



## Pre-Flight - Initiate System - 1 to 2 Race Tracks - Steady State

<p><b>Intelligent Filter Environmental Processor</b></p>	<p>3- Define initial settings and performance measure thresholds</p>	<p>8- Execute Non-STAP Algorithm - Compute No of Secondary Rings - Run NHD - Compute Beam Performance, Determine Null Weights - Determine STAP feasibility</p>	<p>13- Compute Number of Sec. Rings, Run NHD, Compute Beam Performance Measures, Set Nulls, Determine When and Where STAP is Feasible - Evolve Rules</p>	<p>18- Measure Performances - Change Rule Set Accordingly</p>
<p><b>Intelligent Detector Environmental Processor</b></p>	<p>4- Define initial settings and Thresholds for Pfa</p>	<p>9- Compute and Adjust Thresholds for Pfa</p>	<p>14- Compute Detections - Re-compute and Adjust Pfa Thresholds - Evolve Rules</p>	<p>19- Measure Performances - Change Rule Set Accordingly</p>
<p><b>Intelligent Tracker Environmental Processor</b></p>	<p>5- Locate all Potential Discretes, Clutter Boundaries, Shadow Regions, Jammers, Obstacles - Define initial settings and performance measure thresholds</p>	<p>10- Initiate Tracks - Compute Performance Measures (Number of Correct Tracks, Number of Dropped Tracks, Number of Incorrect Tracks)</p>	<p>15- Correlate FAA Data with Tracks - Compute Performance Measures - Number of Tracks, Number of Dropped Tracks, Number of Incorrect Tracks - Evolve Rules</p>	<p>20- Measure Performances - Change Rule Set Accordingly</p>



# Summary/Future Plans

- Introduction/Background
- Approach
- 200 meter NLCD (national land cover data) data – injected targets
- 30 meter NLCD data – moving target simulator (MTS) targets
- Seeing what the sensor sees
- 10 meter DEM data
- Agile Intelligent Radar System (AIRS)
- Future Plans
  - DEM model development and test in SPEAR facility
  - Continue with image map development – seeing what the sensor sees
  - Integrate NLCD, DEM, and DLG data
  - Extend KB control logic design for the total processing chain - AIRS
- Acknowledgements



# References

- D. Weiner, G. Capraro, C. Capraro, M. Wicks, "An Approach for Utilizing Known Terrain and Land Feature Data in Estimation of the Clutter Covariance Matrix," *Proceedings of the 1998 IEEE National Radar Conference*, pp. 381-386, Dallas, TX, May 1998.
- C. Capraro, G. Capraro, D. Weiner, M. Wicks, "Knowledge Based Map Space Time Adaptive Processing (KBMapSTAP)," *Proceedings of the International Conference on Imaging Science, Systems, and Technology*, Vol. 2, pp. 533-538, Las Vegas, NV, June 2001.
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- G. Capraro, M. Wicks, "An Airborne Intelligent Radar System (AIRS), Radar 2004 International Conference on Radar Systems, France
- A. Farina, H. Griffiths, G. Capraro, and M. Wicks, "Knowledge-Based Radar Signal & Data Processing", NATO RTO Lecture Series 233, November 2003



# Acknowledgements

- Dr. Donald Weiner – Syracuse University
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- Mr. William Baldygo – AFRL/SNRT
- Mr. Mark Novak – AFRL/SNRT
- Dr. Michael Wicks – AFRL/SN



# Supplemental Slides



# Performance Measure

## Modified Sample Matrix Inversion

$$\text{MSMI}_i = \frac{\left| s^H \hat{R}^{-1} x_i \right|^2}{s^H \hat{R}^{-1} s}$$

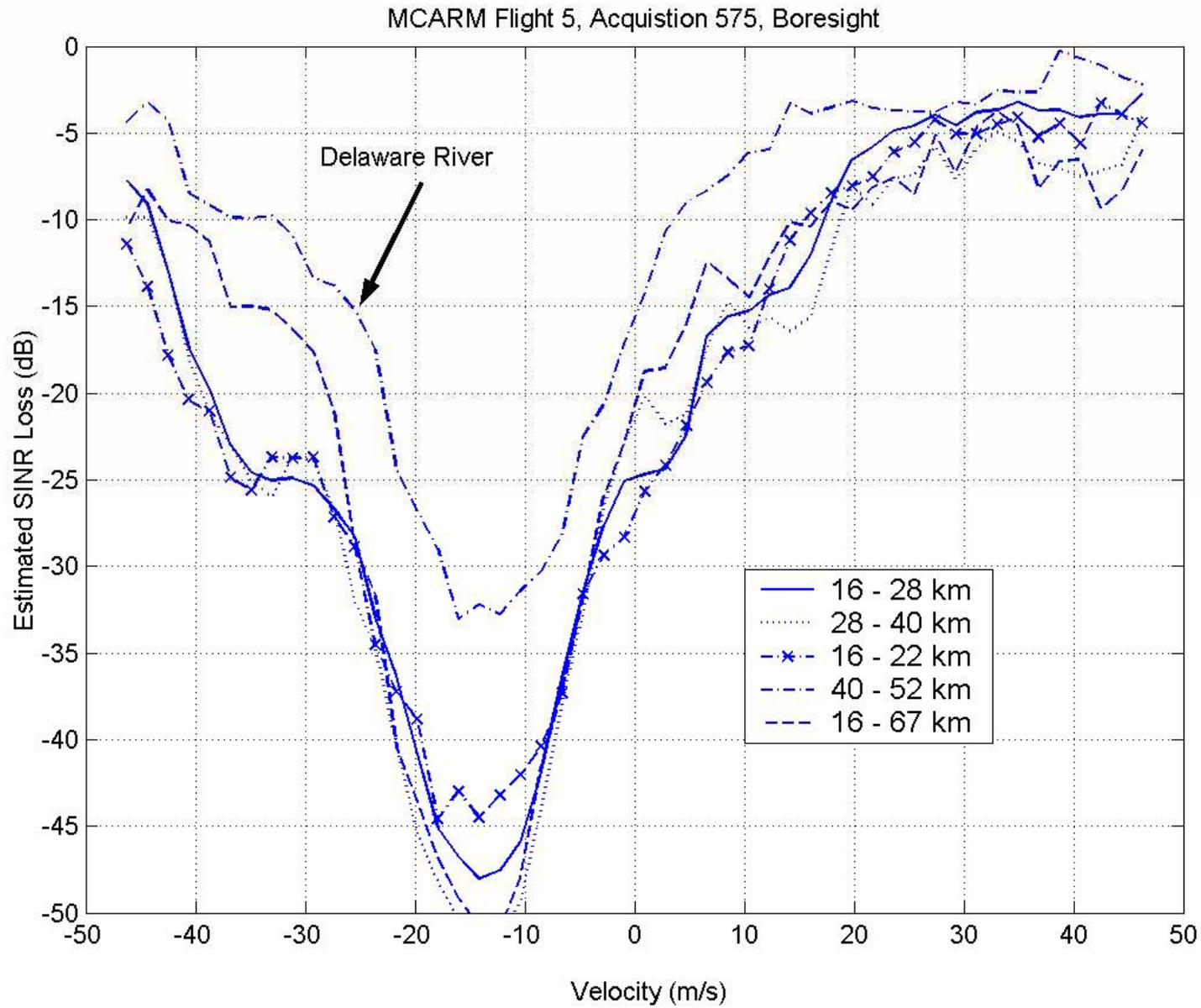
where:  $\hat{R}$  is an estimate of the clutter covariance matrix,  
 $s$  is the steering vector,

$s^H$  is the complex conjugate transpose of the steering vector,

$x_i$  is the radar return vector from the  $i$ th range ring,

$\hat{R}^{-1}$  is the inverse of the estimate of the clutter covariance matrix.

**MSMI is computed for each range ring -  
has a thresholding or detection quality, similar  
to CFAR.**



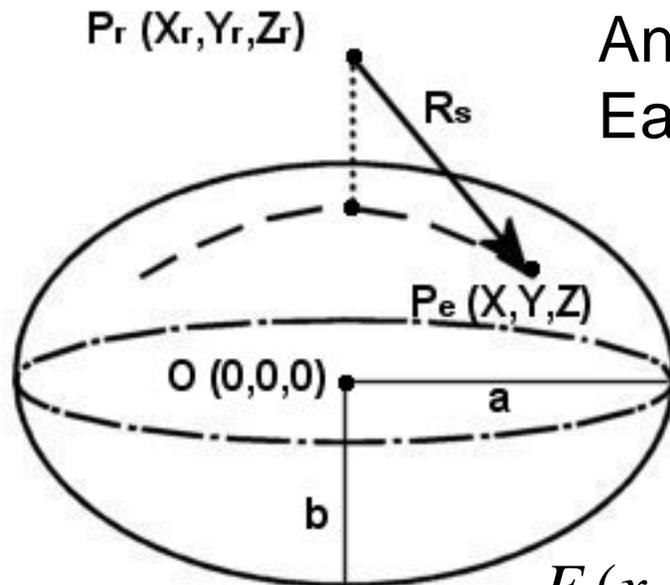


In estimating the covariance matrix, if one wishes to maintain an **average** loss, compared to the optimum, of better than one-half (less than 3 dB), at least  $2N$  samples of data are needed.

I. S. Reed, J. D. Mallet, and L. E. Brennan, "Rapid Convergence Rate in Adaptive Arrays," *IEEE Transactions on Aerospace and Electronic Systems*, Vol. 10, No. 6, pp. 853-863, November, 1974.

- In order to demonstrate that knowledge-aided approaches improve STAP performance we need to use real radar data since similar knowledge sources are used to produce simulated radar data.
- If we use real radar data we have no way of knowing what the true covariance matrices are and, therefore, a reliable performance metric is hard to obtain.
- We need to develop a statistical performance measure, when using real radar data, to evaluate STAP algorithms properly. A couple of data points are not sufficient to prove that new approaches are better than current ones.
- However, a statistical performance measure would require a large collection of radar data with accurate truth information about embedded targets.

# Registration



An oblate spheroid (elliptical) model of the Earth was used.

Newton-Raphson Iterative Method of solution used to solve registration equations.

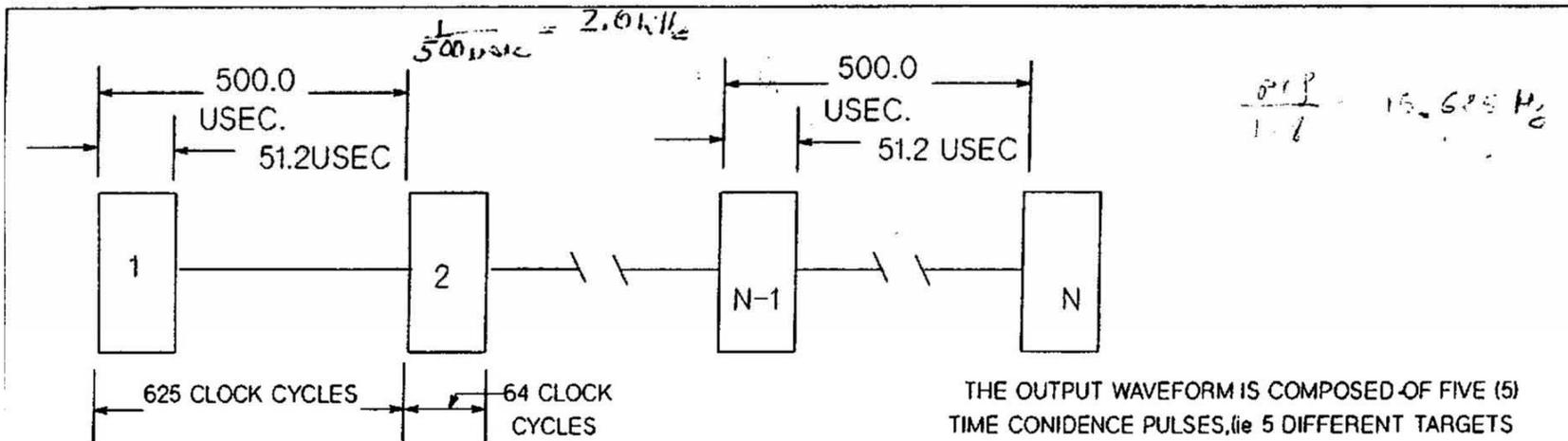
$$F_1(x, y, z) = (x - x_r)^2 + (y - y_r)^2 + (z - z_r)^2 - R_s^2 = 0$$

$$F_2(x, y, z) = (x - x_r)v_{rx} + (y - y_r)v_{ry} + (z - z_r)v_{rz} - \frac{f_d \lambda R_s}{2} = 0$$

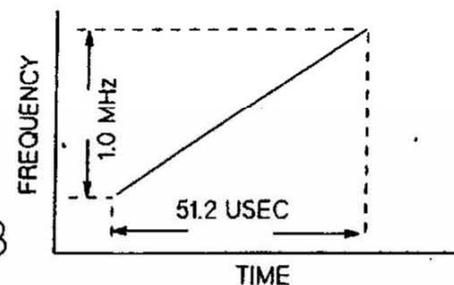
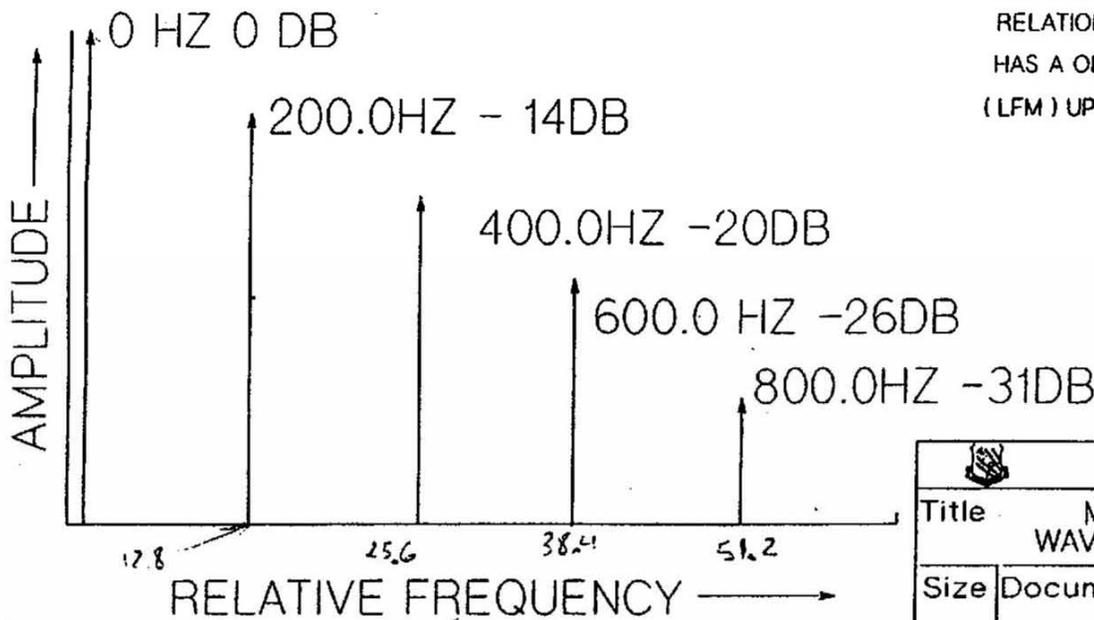
$$F_3(x, y, z) = \frac{x^2}{a^2} + \frac{y^2}{a^2} + \frac{z^2}{b^2} - 1 = 0$$



Monosbatic MTS tones



THE OUTPUT WAVEFORM IS COMPOSED OF FIVE (5) TIME COINCIDENCE PULSES, (ie 5 DIFFERENT TARGETS IN THE SAME RANGE CELL) HAVING THE FREQUENCY AND AMPLITUDE, (ie RELATIVE DOPPLER AND RCS) RELATIONSHIP SHOWN AT THE LEFT. EACH PULSE ALSO HAS A ONE (1.0) MHz LINEAR FREQUENCY MODULATION (LFM) UP CHIRP. (SEE DIAGRAM BELOW)

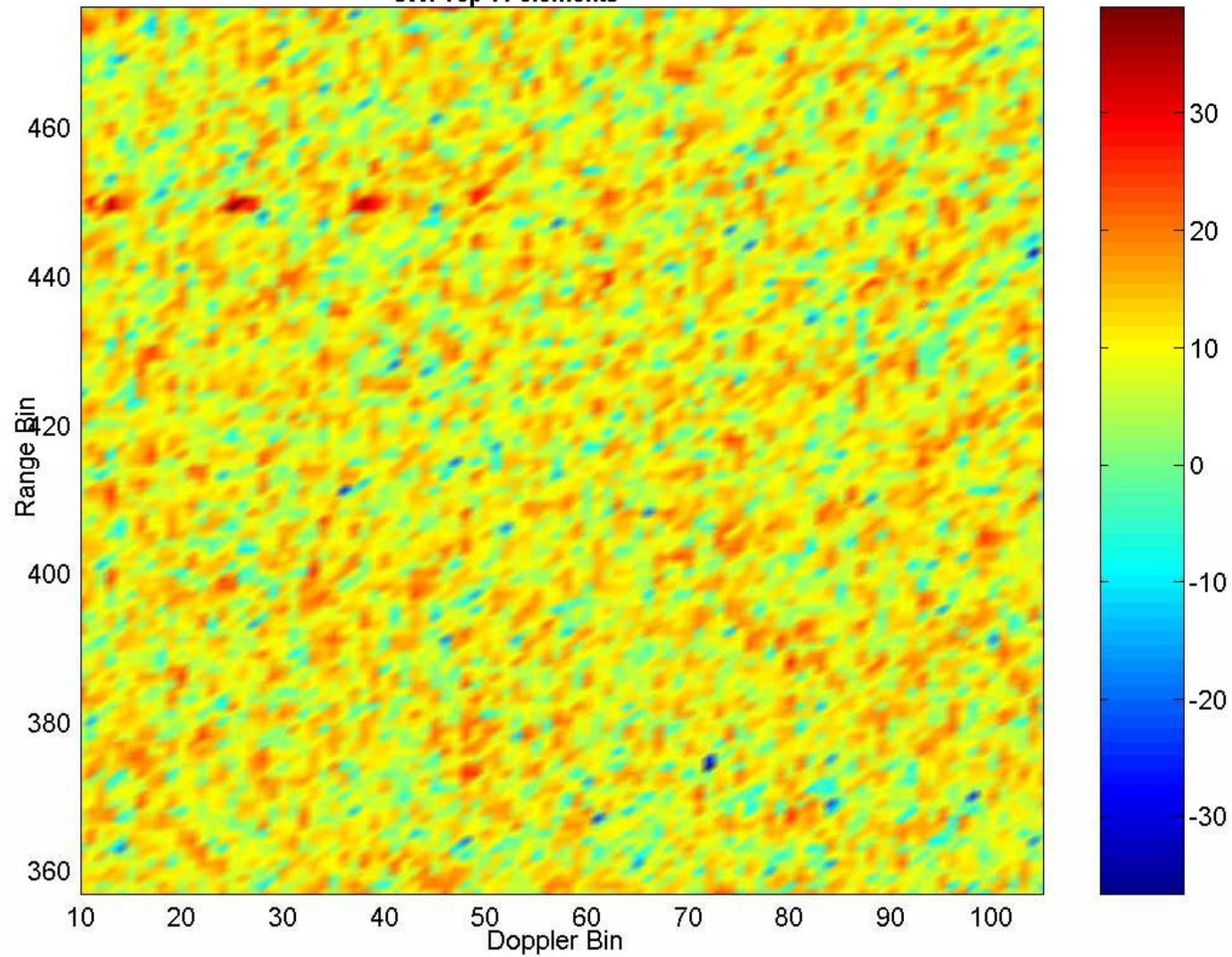


0 HZ RELATIVE FREQUENCY CORRESPONDS TO 1240.000000 MHz

Rome Laboratory		
Title: MOVING TARGET SIMULATOR WAVEFORM PARAMETERS MCARM		
Size	Document Number MTS7.GAL	REV
Date: 2 JUNE 1995		Sheet 7 of 8

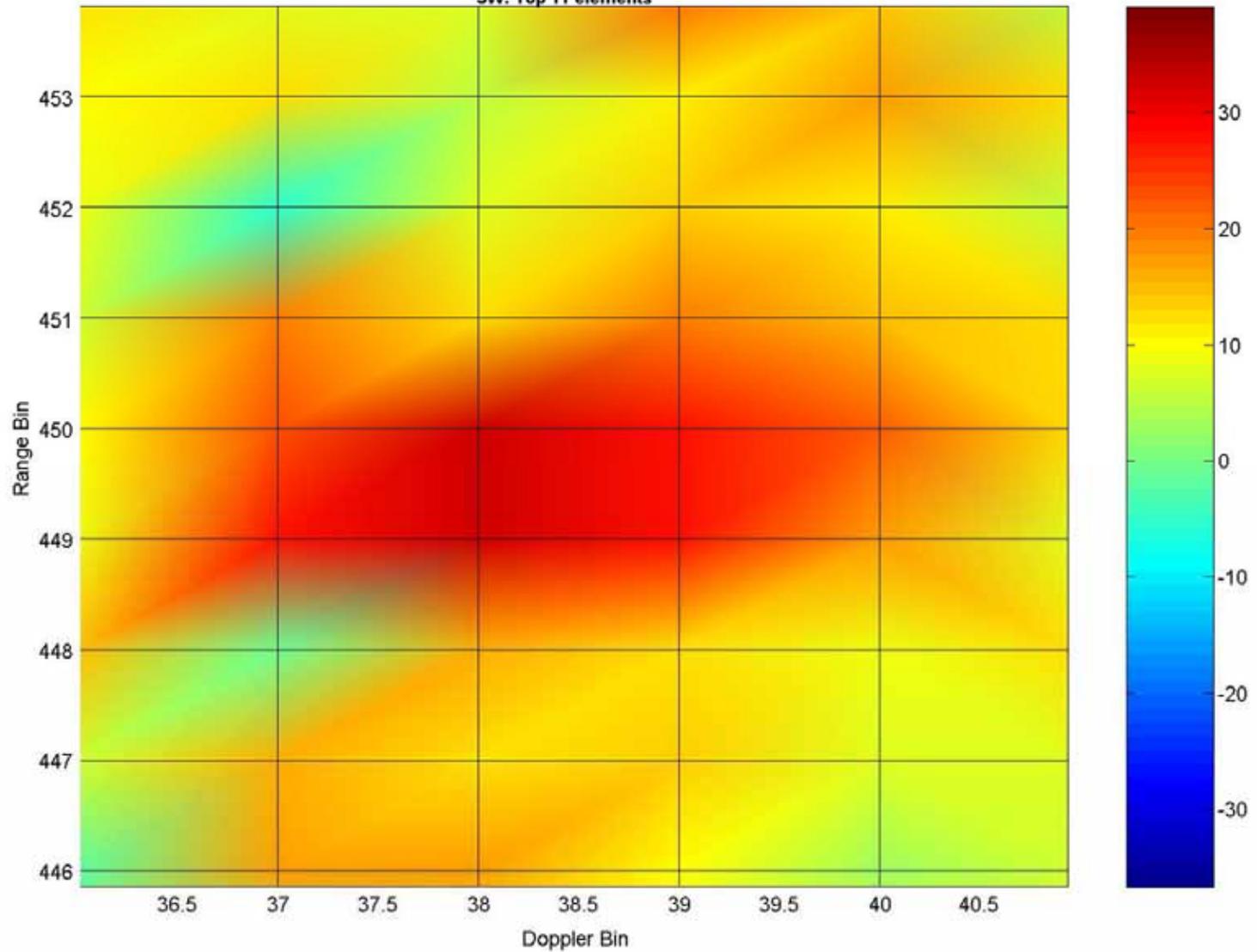


MCARM Flight 5 Acq 152  
MSMI: mean=13.1148 std=15.9819 ang=85  
SW: Top 11 elements





MCARM Flight 5 Acq 152  
MSMI: mean=13.1148 std=15.9819 ang=85  
-400Hz MTS Chirp  
SW: Top 11 elements





# Bac 1-11





Antenna Main Beam

Assuming No  
Target Spreading

N  
N-1



Interfering Doppler

Guard Ring  
Test Cell  
Guard Ring

Test Ring

N/2

Interfering Patch

16  
15

Radar

2  
1



capraro  
technologies  
inc.



Antenna Main Beam

Accounting For Target Spreading

Guard Ring (s)  
Selected Training Ring  
Guard Ring (s)

Guard Ring  
Test Cell  
Guard Ring

Test Ring

Interfering Patch

N/2

16  
15

Radar

2  
1



capraro technologies inc.

